

SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : LANDING/DECELERATION-LGC FMEA NO 02-1A -010 -1 REV:09/19/88

ASSEMBLY : MAIN LANDING GEAR (MLG)	CRIT. FUNC:	1
P/N RI : MC621-0011	CRIT. HDW:	1
P/N VENDOR: 1170353 MENASCO	VEHICLE	102 103 104
QUANTITY : 2	EFFECTIVITY:	X X X
: ONE LEFT MAIN GEAR	PHASE(S):	PL LO OO DO LS X
: ONE RIGHT MAIN GEAR		

PREPARED BY:	DES	R. A. GORDON	APPROVED BY:	DES	<i>R.A. Gordon 9/21/88</i>	REDUNDANCY SCREEN: A-	B-	C-
REL	J. S. MULLEN	REL	<i>J.S. Mullen</i>	SSM	<i>Charles Campbell</i>	APPROVED BY (NASA):		
QE	W. J. SMITH	QE	<i>W.J. Smith</i>	REL	<i>W.J. Smith 9/27</i>			

ITEM:
MAIN LANDING GEAR TORQUE TUBE ASSEMBLY

FUNCTION:
PROVIDES SUPPORT AND DRIVE TO LOCK BRACE ASSEMBLY FROM THE HYDRAULIC ACTUATOR IN RELEASE OF MLG STRUTS, ALSO PROVIDES SUPPORT BETWEEN TRUNNION ASSEMBLIES.

FAILURE MODE:
STRUCTURAL FAILURE

CAUSE(S):
OVERLOAD, DEFECTIVE PART/MATERIAL.

EFFECT(S) ON:

- (A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE
- (A) LOSS OF LOAD CARRYING CAPABILITY.
- (B) DAMAGE TO VEHICLE STRUCTURE.
- (C,D) PROBABLE LOSS OF MISSION/CREW/VEHICLE DUE TO GEAR COLLAPSE.

DISPOSITION & RATIONALE:
(A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE

(A) DESIGN

DESIGNED TO FATIGUE LOAD SPECTRUM FOR LANDING, TAXI, AND GROUND HANDLING CONDITIONS. DESIGNED TO LANDING IMPACT LOADS (SPIN-UP AND SPRING BACK INCLUDING CROSSWIND DRIFT CONDITIONS) USING A MINIMUM FACTOR OF SAFETY OF 1.0 TO YIELD STRENGTH OF MATERIAL IN ACCORDANCE WITH ESTABLISHED CRITERIA FOR COMMERCIAL AND MILITARY AIRCRAFT. DESIGNED TO A MINIMUM FACTOR OF SAFETY OF 1.4 FOR TAXI AND GROUND HANDLING LOADS MATERIAL PROCESSES - BARE PARTS ARE NOT EXPOSED TO CORROSIVE ENVIRONMENT IN PLATING SHOP MORE THAN 30 DAYS AND PARTS ARE SHOT PEENED AFTER MACHINE OPERATIONS TO PREVENT STRESS CORROSION ON 300 M MATERIALS.

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(B) TEST

QUALIFICATION TESTS: CERTIFICATION INCLUDES ULTIMATE STRENGTH TEST, SHOCK STRUT DROP TESTS, STATIC LOADS TEST, DYNAMIC TESTS AND 400 DEPLOYMENT CYCLES.

THE TORQUE TUBE ASSEMBLY WAS CERTIFIED AS AN INTEGRAL PART OF THE MLG MECHANISM INSTALLATION (LANDING GEAR OPERATION) - 32 CYCLES OF THE LANDING GEAR RISING ALT, 15 DEVELOPMENT CYCLES AND 353 QUALIFICATION LI CYCLES FOR A TOTAL OF 400 CYCLES. (THE LANDING GEAR WAS CYCLED FROM UP AND LOCKED TO DOWN AND LOCKED EACH TIME).

ENVIRONMENT:

HIGH TEMP TESTS; 3 CYCLES AT 140 DEG F

COLD TEMP TESTS; 3 CYCLES AT -35 DEG F TO -40 DEG F

THE TORQUE TUBE WAS ALSO TESTED AS AN INTEGRAL PART OF THE MLG SHOCK STRUT ASSEMBLY DURING DROP TESTS - ELEVEN DROP TESTS WERE PERFORMED TO SATISFY THE DESIGN REQUIREMENTS FOR THE SHOCK STRUT ASSEMBLY.

MAXIMUM VERTICAL LOAD WAS 179,817 LBS.

MAXIMUM SINK SPEED WAS 11.69 FPS.

FATIGUE LOAD SPECTRUM TESTS WERE CONDUCTED FOR LANDING, LANDING ROLLOUT BRAKING AND TURNING LOAD CONDITIONS - THE STRUT WAS SUBJECTED TO CYCLIC APPLICATION OF VERTICAL, FORE/AFT AND SIDE LOADS IN EACH CONDITION.

ACCEPTANCE TESTS: ACCEPTANCE INCLUDES VERIFICATION THAT CERTIFIED MATERIALS AND PROCESSES WERE USED. ACCEPTANCE TESTS ALSO VERIFY DIMENSIONS, WEIGHTS AND FINISHES.

OMRSD: MLG ZONAL DETAIL VISUAL INSPECTION; THE TORQUE TUBE AND IT'S ATTACHMENTS ARE INSPECTED FOR CONDITION AND SECURITY.

FREQUENCY - ALL VEHICLES AT GROUND TURNAROUND.

(C) INSPECTION

RECEIVING INSPECTION

INSPECTION VERIFIES ALL RAW MATERIALS TO COMPLY WITH MATERIAL REQUIREMENTS THROUGH PERIODIC COUPON ANALYSIS.

CONTAMINATION CONTROL

ALL CLEANLINESS LEVELS VERIFIED BY INSPECTION.

ASSEMBLY/INSTALLATION

ALL MATERIAL PROCESSES VERIFIED BY MIPS PRIOR TO NEXT MANUFACTURING OPERATIONS.

NONDESTRUCTIVE EVALUATION

MATERIAL SURFACE DEFECTS ARE VERIFIED BY MAGNETIC PARTICLE, NITAL ETCH, AND ULTRASONIC INSPECTION.

CRITICAL PROCESSES

HEAT TREATING, SHOT PEENING, CHROME AND CD-TI PLATING VERIFIED BY INSPECTION.

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TESTING

TORSIONAL OVERLOADS ARE VERIFIED BY DYNAMIC AND STATIC TESTS PERFORMED DURING QUALIFICATION TESTING.

HANDLING/PACKAGING

HANDLING AND PACKAGING REQUIREMENTS VERIFIED BY INSPECTION.

(D) FAILURE HISTORY

NONE

(E) OPERATIONAL USE

NONE

02-1A-24