

SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : LANDING DECELERATION FMEA NO 02-1F -G09-TO-F01 REV:06/27/

ASSEMBLY : MLG STRUT ACTUATOR
 P/N RI : MC287-0034
 P/N VENDOR : PARKER-BERTEA
 QUANTITY : 2
 : ONE PER ACTUATOR
 :

	VEHICLE	102	103	104
	EFFECTIVITY:	X	X	X
	PHASE(S):	PL LO	OO	DO X LS

CRIT. FUNC:

CRIT. HDW:

PREPARED BY: DES N LEVERT
 REL C NELSON
 QE M SAVALA

REDUNDANCY SCREEN: A-N/A B-N/A C-N/
 APPROVED BY: DES [Signature]
 REL [Signature]
 QE [Signature]

APPROVED BY (NASA):
 SSM [Signature]
 REL [Signature]
 QE [Signature]

ITEM:
 TIMING ORIFICE

FUNCTION:
 CONTROLS HYDRAULIC FLUID INTERCHANGE FROM ONE SIDE OF PISTON TO OTHER
 ACHIEVE PROPER GEAR DEPLOYMENT TIME.

FAILURE MODE:
 BLOCKED

CAUSE(S):
 CONTAMINATION

EFFECT(S) ON:
 (A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE

(A, B) LANDING GEAR WILL NOT DEPLOY DUE TO HYDRAULIC LOCKUP OR WILL BE
 INCREASED EXTENSION TIME.

(C) NONE, COMMITTED TO LANDING.

(D) POSSIBLE LOSS OF CREW/VEHICLE IF GEAR DOES NOT DEPLOY.

DISPOSITION & RATIONALE:
 (A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE

(A) DESIGN
 THE ORIFICE IS PROTECTED BY 80 MICRON FILTERS IN THE ACTUATOR CONTROL
 VALVE MODULE EXTEND AND RETRACT PASSAGES AND A 5 MICRON SYSTEM SUPPLY
 FILTER. SILTING IS UNLIKELY DUE TO THE ORIFICE SIZE WHICH RANGES FROM
 0.163 TO 0.268 INCH FOR THE MAIN LANDING GEAR.

(B) TEST
 QUALIFICATION-RANDOM VIBRATION AND ENDURANCE TESTS REPRESENTATIVE
 MISSION ENVIRONMENT. ACTUATOR EXTEND TIME TEST, PROOF PRESSURE TEST
 POST TEST PROCEDURE INCLUDES DISASSEMBLY AND INSPECTION OF WORKING
 COMPONENTS.

ACCEPTANCE-ACTUATOR EXTEND TIME TEST, UNIT CLEANLINESS TEST.

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SUBSYSTEM :LANDING DECELERATION FMEA NO 02-1F -G09-TO-F01 REV:06/27/8

OMRSD-EXTEND--LEFT AND RIGHT MAIN LANDING GEAR FROM CREW STATIC PERFORMED PRIOR TO EACH MISSION. POST LANDING HYDRAULIC RESERVOIR EFFLUENT SAMPLES, PERFORMED AFTER EVERY FLIGHT; VERIFY THAT RESULTS FLUID SAMPLE CONTAMINATION MEET SPECIFICATION. GENERAL REQUIREMENT 5. VERIFY ALL HYDRAULIC FLUID USED TO SERVICE VEHICLE IS PER MIL-H-83282.

(C) INSPECTION

RECEIVING INSPECTION

CERTIFICATION RECORDS AND CERTIFIED TEST REPORTS ARE MAINTAINED WITH CERTIFYING MATERIAL AND PHYSICAL PROPERTIES.

CONTAMINATION CONTROL

SUPPLIER TEST STAND FLUID PARTICLE COUNT CHECKED TWICE A DAY, WHEN APPLICABLE. FLUID CONTAMINATION PARTICLE COUNT CONDUCTED PRIOR TO ATP AFTER ATP, A FLUID SAMPLE IS DRAWN TO VERIFY FLUID CLEANLINESS. IF CONTAMINATED, ACTUATOR IS CYCLED AND FLUSHED UNTIL CONFIRMATION IS ATTAINED. SYSTEM CLEANLINESS IS VERIFIED TO LEVEL 220 PER MA0110-301.

CRITICAL PROCESSES

HEAT TREAT PROCESS IS VERIFIED BY INSPECTION.

NDE

INSPECTION VERIFIES THAT DETAIL PARTS ARE MAGNETIC PARTICLE OR PENETRANT INSPECTED, DEPENDING ON THE ALLOY.

ASSEMBLY/INSTALLATION

INSPECTION OF DIMENSIONS AT FINAL INSPECTION. COMPONENT PARTS VERIFIED UNDAMAGED PRIOR TO CLEANING AND PACKAGING.

TESTING

ATP IS VERIFIED BY INSPECTION.

HANDLING/PACKAGING

PARTS PROTECTION TO PRECLUDE CONTAMINATION DURING SHIPMENT IS VERIFIED BY INSPECTION.

(D) FAILURE HISTORY

THERE IS NO HISTORY OF FAILURE FOR THIS FAILURE MODE.

(E) OPERATIONAL USE

NONE