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PRINT DATE: 09/01/93

**FAILURE MODES EFFECTS ANALYSIS (FMEA) - CRITICAL HARDWARE
NUMBER: 05-6N-2040-X**

SUBSYSTEM NAME: EPD&C - AUXILIARY POWER UNIT

REVISION: 1 08/30/93

	PART NAME VENDOR NAME	PART NUMBER VENDOR NUMBER
LRU	: PANEL R2	V070-730277
SRU	: FUSE	ME451-0009-1001

PART DATA

**EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:
FUSE (1 AMP) - AUXILIARY POWER UNIT (APU) AUTO SHUTDOWN INHIBIT CONTROL
CIRCUIT**

**REFERENCE DESIGNATORS: 32V73A2F44
32V73A2F45
32V73A2F93
32V73A2F97
32V73A2F98
32V73A2F100**

**QUANTITY OF LIKE ITEMS: 6
SIX**

**FUNCTION:
PROVIDES OVERCURRENT PROTECTION FOR APU AUTO SHUTDOWN INHIBIT
CONTROL CIRCUIT.**

**FAILURE MODES EFFECTS ANALYSIS (FMEA) - CRITICAL FAILURE MODE
NUMBER: 05-6N-2040-01**

REVISION# 1 06/30/93

SUBSYSTEM NAME: EPD&C - AUXILIARY POWER UNIT

LRU: PANEL R2

ITEM NAME: FUSE

CRITICALITY OF THIS
FAILUREMODE: 1R3

FAILURE MODE:

FAILS OPEN, FAILS TO CONDUCT

MISSION PHASE:

PL	PRELAUNCH
LO	LIFT-OFF
DO	DE-ORBIT
LS	LANDING SAFING

VEHICLE/PAYLOAD/KIT EFFECTIVITY:

102	COLUMBIA
103	DISCOVERY
104	ATLANTIS
105	ENDEAVOUR

CAUSE:

STRUCTURAL FAILURE, CONTAMINATION, VIBRATION, MECHANICAL SHOCK,
PROCESSING ANOMALY, THERMAL STRESS

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN

A) PASS
B) FAIL
C) PASS

PASS/FAIL RATIONALE:

A)

B)

FIRST FAILURE NOT DETECTABLE IN FLIGHT SINCE THE OPERATIONAL STATUS OF
THIS FUSE IS NOT MONITORED.

C)

- FAILURE EFFECTS -

(A) SUBSYSTEM:

LOSS OF REDUNDANCY TO INHIBIT AUTOMATIC SHUTDOWN FROM AN
OVERSPEED/UNDERSPEED CONDITION.

(B) INTERFACING SUBSYSTEM(S):

NO EFFECT - FIRST FAILURE. REDUNDANT CIRCUIT WILL PROVIDE APU AUTO
SHUTDOWN INHIBIT COMMAND.

(C) MISSION:

**FAILURE MODES EFFECTS ANALYSIS (FMEA) - CRITICAL FAILURE MODE
NUMBER: 05-6N-2040-01**

NO EFFECT - FIRST FAILURE. ABORT DECISION REQUIRED AFTER THREE FAILURES (LOSS OF ONE APU).

(D) CREW, VEHICLE, AND ELEMENT(S):
NO EFFECT - FIRST FAILURE

(E) FUNCTIONAL CRITICALITY EFFECTS:
POSSIBLE LOSS OF CREW/VEHICLE AFTER THREE OTHER FAILURES (FUSE OPENS, FALSE OVERSPEED/UNDERSPEED INDICATION ON APU, LOSS OF SECOND APU) DUE TO LOSS OF TWO OF THREE APUS.

-DISPOSITION RATIONALE-

(A) DESIGN:
REFER TO APPENDIX D, ITEM NO. 2 - FUSE, AXIAL LEAD/CARTRIDGE

(B) TEST:
REFER TO APPENDIX D, ITEM NO. 2 - FUSE, AXIAL LEAD/CARTRIDGE

GROUND TURNAROUND TEST - APU 1/2/3 CONTROLLER TEST THROUGH GROUND CONNECTION PERFORMED EVERY FLOW OR AFTER LRU RETEST OF APU ASSEMBLY, AFTER LRU RETEST OF CONTROLLER ASSEMBLY OR AFTER CIG RETEST.

(C) INSPECTION:
REFER TO APPENDIX D, ITEM NO. 2 - FUSE, AXIAL LEAD/CARTRIDGE

(D) FAILURE HISTORY:
REFER TO APPENDIX D, ITEM NO. 2 - FUSE, AXIAL LEAD/CARTRIDGE

(E) OPERATIONAL USE:
NONE

- APPROVALS -

EDITORIALLY APPROVED : RI
EDITORIALLY APPROVED : JSC
TECHNICAL APPROVAL : VIA CR

Handwritten signatures and dates:
- [Signature] 9/1/93
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- S50370