PAGE: 1 PRINT DATE: 01/24/96

FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CIL HARDWARE NUMBER:05-60-200606 -X

SUBSYSTEM NAME: EPD&C - GN&C

REVISION: 0

06/28/88

PART DATA

PART NAME VENDOR NAME

PART NUMBER VENDOR NUMBER

LRU

:PANEL 017

V070-730397

SRU

:SWITCH, TOGGLE

ME452-0102-7301

**EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:** 

SWITCH, ATVC POWER, 3PST

REFERENCE DESIGNATORS:

33V73A17\$1

33V73A17S2

33V73A17\$3 33V73A17S4

QUANTITY OF LIKE ITEMS:

FOUR REQUIRED

FUNCTION:

PROVIDES CAPABILITY FOR SWITCHING ATVC POWER ON AND OFF, INCLUDING MPS ISOLATION VALVE DRIVER POWER.

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FAILURE MODES EFFECTS	ANALYSIS FMEA	- CIL FAILURE MODE
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NUMBER: 05-60-200606-01

REVISION#; 1

01/22/98

SUBSYSTEM NAME: EPD&C-GUIDANCE, NAVIGATION, & CONTROL (05-1)

CRITICALITY OF THIS

LRU: PANEL 017

ITEM NAME: SWITCH, TOGGLE

FAILURE MODE: 1R2

#### FAILURE MODE:

FAILS TO TRANSFER FROM OFF, 3 POLES OPEN OR SHORTED TO GROUND.

MISSION PRASE:

LO LIFT-OFF

VEHICLE/PAYLOAD/KIT EFFECTIVITY:

102 COLUMBIA

103 DISCOVERY

104 **ATLANTIS** 

105 ENDEAVOUR

#### CAUSE:

MECHANICAL SHOCK, VIBRATION, CONTAMINATION, PIECE PART STRUCTURAL FAILURE, PROCESSING ANOMALY,

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN

A) PASS

B) PASS

C) PASS

## PASS/FAIL RATIONALE:

A)

B)

C)

- FAILURE EFFECTS .

#### (A) SUBSYSTEM:

LOSS OF CONTROL POWER TO ONE OF FOUR ATVC'S AND ASSOCIATED MPS ISQLATION VALVE DRIVER.

## 90.17 (B) INTERFACING SUBSYSTEM(S):

LOSS OF ONE OF FOUR ATVC'S AND ASSOCIATED MPS ISOLATION VALVE DRIVER.

(C) MISSION:

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# FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CIL FAILURE MODE

NUMBER: 05-60-200606-01

NO EFFECT

### (D) CREW, VEHICLE, AND ELEMENT(S):

FIRST FAILURE NO EFFECT. SECOND FAILURE (LOSS OF ADDITIONAL ATVC AND ASSOCIATED MPS ISOLATION VALVE DRIVER DUE TO A SECOND SWITCH FAILURE) CAUSES A TWO ON TWO FORCE FIGHT CONDITION FOR MPS TVC. FLIGHT CONTROL SUBSYSTEM CAN NOT TOLERATE THIS CONDITION.

### (E) FUNCTIONAL CRITICALITY EFFECTS:

CRITICALITY 1R BECAUSE LOSS OF MPS AND SRB THRUST VECTOR CONTROL MAY CAUSE LOSS OF CREWIVEHICLE. SINGLE POLE (ISOLATION VALVE DRIVER POWER) SWITCH FAILURE IS CRITICALITY 1R3 AND UNDETECTABLE IN FLIGHT.

#### -DISPOSITION RATIONALE-

(A) DESIGN:

REFER TO APPENDIX A, ITEM #1 - TOGGLE SWITCH,

(B) TEST:

REFER TO APPENDIX A, ITEM #1 - TOGGLE SWITCH.

GROUND TURNAROUND TEST

SWITCH FUNCTIONS ARE VERIFIED DURING GROUND TURNAROUND TESTING.

(C) INSPECTION:

REFER TO APPENDIX A, ITEM #1 - TOGGLE SWITCH.

(D) FAILURE HISTORY;

RÉFER TO APPENDIX A, ITEM #1 - TOGGLE SWITCH.

(E) OPERATIONAL USE:

NONE

- APPROVALS -

EDITORIALLY APPROVED

EDITORIALLY APPROVED

: RI

TECHNICAL APPROVAL

: JSC : APPROVAL FORM Sam Seoury 2-12-96

95-CIL-004/RI