FAILURE MODES EFFECTS ANALYSIS (FMEA) - NON-CIL HARDWARE NUMBER:05-60-GN0804 -X

SUBSYSTEM NAME: EPD&C-GUIDANCE, NAVIGATION, & CONTROL (05-1)

REVISION:	1	01/22/96
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	PA	RT DATA
	PART NAME	PART NUMBER
	VENDOR NAME	VENDOR NUMBER
LRIJ	:AFT PCA 3	V070-765240
LRU	:AFT PCA 4, 5, 6	V070-765280
LRU	:AFT PCA 3	V070-765330
SRU	:DIODE	JANTX1N1204RA

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS: DIODE 12 AMPS

REFERENCE DESIGNATORS:

54V76A134A3CR6 54V76A134A3CR7 56V76A133A3CR9 56V76A133A3CR10

QUANTITY OF LIKE ITEMS: FOUR-TWO PER MAIN BUS

FUNCTION:

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PERMITS ONE WAY CONDUCTION AND PROVIDES MAIN BUS ISOLATION OF ELECTRICAL CURRENT FROM ORBITER MAIN BUS A AND C THROUGH RPC'S TO SRB RGA 3 AND 4.

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PRINT DATE: 01/24/96

FAILURE MODES EFFECTS	ANALYSIS FN	1EA –	NON-CIL FAILURE MODE NUMBER: 05-60-GN08	04-01		
SUBSYSTEM NAME: EPD&C LRU: AFT PCA 3, 4 ITEM NAME: DIODE	-GUIDANCE,	NAVIG	REVISION#: 1 FATION, & CONTROL (05-1) CRITICALIT FAILURE MI			
FAILURE MODE: OPEN, FAILS TO CONDUCT.						
MISSION PHASE: LC	UFT-OFF					
VEHICLE/PAYLOAD/KIT EFF	ECTIVITY:	103	COLUMBIA DISCOVERY ATLANTIS ENDEAVOUR			
CAUSE: THERMAL STRESS, VIBRATIO PROCESSING ANOMALY.	ON, MECHAN	ICAL S	TRESS, ELECTRICAL STRE	ss,		
CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO						
REDUNDANCY SCREEN	A) PASS B) N/A C) PASS		***			
PASS/FAIL RATIONALE: A)						
B) B SCREEN NOT APPLICABLE DUE TO REDUNDANT POWER PATHS TO RGA'S, LOSS OF ANY OF THE FOUR RGA'S IS READILY APPARENT DURING FLIGHT USE.						
C}			•			
CORRECTING ACTION: NO	NE:					
CORRECTING ACTION DESC	RIPTION:					

(A) SUBSYSTEM:

编织

- FAILURE EFFECTS -

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FAILURE MODES EFFECTS ANALYSIS (FMEA) - NON-CIL FAILURE MODE NUMBER: 05-60-600804-01

LOSS OF REDUNDANT POWER TO ONE RGA.

(B) INTERFACING SUBSYSTEM(S):

NO EFFECT. RGA STILL HAS REDUNDANT POWER PATH.

(C) MISSIQN:

NO EFFECT.

(D) CREW, VEHICLE, AND ELEMENT(S):

NO EFFECT FIRST FAILURE. SECOND FAILURE (LOSS OF ASSOCIATED REDUNDANT POWER PATH) RESULTS IN LOSS OF TWO OF FOUR RGA'S. REDUNDANCY MANAGEMENT TOLERATES THIS CONDITION. THIRD FAILURE (LOSS OF A THIRD RGA) RESULTS IN LOSS OF VEHICLE AS REDUNDANCY MANAGEMENT CAN NOT TOLERATE THIS CONDITION.

(E) FUNCTIONAL CRITICALITY EFFECTS:

CRITICALITY 1R BECAUSE LOSS OF RATE FEEDBACK INFORMATION MAY CAUSE LOSS OF VEHICLE.

- APPROVALS -

EDITORIALLY APPROVED

V ADDROVED

EDITORIALLY APPROVED TECHNICAL APPROVAL

: RI : JSC

: APPROVAL FORM

95-CIL-004-RI