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PRINT DATE: 06/29/92

FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL HARDWARE

NUMBER: 06-18-0830-X

SUBSYSTEM NAME: ARS - COOLING

REVISION: 7 06/26/92

PART NAME VENDOR NAME

PART NUMBER VENDOR NUMBER

■ LRU ;

REGENERABLE COS REMOVAL SYSTEM MC623-0016

■ SRU :

VALVE, MOTORIZED

SV806920

PART DATA

- EXTENDED DESCRIPTION OF PART UNDER AMALYSIS: VACUUM CYCLE VALVE (VCV)
- QUANTITY OF LIKE ITEMS: 4
- FUNCTION:

THESE VALVES DIRECT THE AIR FLOW INTO THE SORBENT BEDS DURING THE ABSORBTION CYCLE OR EXPOSE THE BED TO THE VACUUM CYCLE DURING BED REGENERATION. THE 4 VACUUM CYCLE VALVES ARE CONNECTED TOGETHER BY TWO INDEPENDENT LINKAGE ASSEMBLIES. THE AIR POPPET OF BED "A" IS CROSS-TIED TO THE VACUUM POPPET OF BED "B" AND VICE-VERSA. EACH LINKAGE SET IS DRIVEN BY AN ACTUATOR WHICH SYNCHRONOUSLY SWINGS THE POPPETS.

PAGE: 6 PRINT DATE: 06/29/92 FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL FAILURE MODE NUMBER: 06-18-0830-02 REVISION# 7 06/26/92 R SUBSYSTEM: ARS - COOLING LRU :REGENERABLE COZ REMOVAL SYSTEM CRITICALITY OF THIS ITEM NAME: VALVE, MOTORIZED FAILURE MODE:2/2 • FAILURE MODE: FAILS CLOSED, RESTRICTED FLOW MISSION PHASE: OO ON-ORBIT ■ VEHICLE/PAYLOAD/KIT EFFECTIVITY: 102 COLUMBIA : 105 ENDEAVOUR CAUSE: MECHANICAL SHOCK, VIBRATION, CORROSION, CONTAMINATION, PHYSICAL BINDING/ JAMMING, BROKEN TORSION SPRING. ■ CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO ■ REDUNDANCY SCREEN A) N/A B) N/A C) N/A PASS/FAIL RATIONALE: **■** A) **⇒** B) ■ C) W MASTER MEAS. LIST NUMBERS: V61P2901A : V61P2902A : V61P2911A : V61P2912A - FAILURE EFFECTS -

■ (A) SUBSYSTEM:
REDUCED AIR FLOW THROUGH THE BEDS. THE RCRS IS NOT REMOVING CO2 FROM CABIN AIR. LOSS OF USE OF THE RCRS.

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- (B) INTERFACING SUBSYSTEM(\$): INCREASED PPCO2 IN CABIN.
- (C) MISSION: POSSIBLE EARLY MISSION TERMINATION.
- (D) CREW, VEHICLE, AND ELEMENT(S): NO EFFECT
- (E) FUNCTIONAL CRITICALITY EFFECTS:
 LOSS OF THE USE OF THE RCRS. THE LICH SUPPLY MUST BE USED FOR CO2
 REMOVAL UNTIL LANDING. THE LICH SUPPLY IS ADEQUATE TO ACCOMMODATE 3
 DAY MISSION. LOSS OF ALL THESE BACKUPS MAY RESULT IN LOSS OF THE
 CREW/VEHICLE. A 1R3 PPP CRITICALITY RESULTS.

- DISPOSITION RATIONALE -

THE RCRS VACUUM CYCLE VALVE DESIGN IS AN ALUMINUM HOUSING VALVE CONTAINING TWO POPPET VALVES, AIR POPPET AND A VACUUM POPPET. THE POPPET SHAFT INCORPORATES A TORSION SPRING DESIGN WHICH PERMITS IMPROPER SHAFT ROTATION AND/OR WHEN THE POPPET DELTA PRESSURE IS GREATER THAN 3.4 PSID, WITHOUT POPPET BEING LIFTED OFF, PREVENTS LINKAGE/VALVE STRUCTURE DAMAGE AND AIR LEAKAGE TO VACUUM. POPPET ARM SHAFT RIDES ON A RULON BUSHING AND IS DRIVEN BY AN ACTUATOR THROUGH HARD LINKAGES. THE POPPET SEAT IS A MOLDED SILICONE RUBBER SEAL. CORROSION RESISTANT MATERIAL USED FOR SPRING FABRICATION.

(B) TEST: QUALIFICATION TEST FOR 100 MISSION LIFE: VCV IS TESTED WHEN INSTALLED AT THE RCRS ASSEMBLY LEVEL. RANDOM VIBRATION INCREASING AT PLUS 6 db/oct FROM 20 TO 45 HZ; CONSTANT AT 0.003 g2/HZ FROM 45 TO 1000 HZ; DECREASING AT MINUS 6 db/oct FROM 1000 TO 2000 HZ FOR 48 MINUTES PER AXIS IN THREE ORTHOGONAL AXES. VALVE IS TESTED FOR 50,000 CYCLES WITH NO EVIDENCE OF FAILURE OR DAMAGE TO VERIFY DESIGN LIFE REQUIREMENTS.

ACCEPTANCE TEST:

FLOW TESTED DURING ACCEPTANCE TESTING TO VERIFY NO FLOW RESTRICTION. PROOF PRESSURE TEST AT 1.5 TIMES THE OPERATING PRESSURE (18PSIA) DIFFERENTIAL AND HOLD FOR 5 MINUTES WITH NO STRUCTURE DAMAGE OR PERMANENT DEFORMATION.

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m (C) INSPECTION:

RECEIVING INSPECTION INCOMING MATERIAL IDENTIFICATION AND CERTIFICATION VERIFIED BY INSPECTION. 100% VISUAL DIMENSIONAL INSPECTION OF SEALS, SPRINGS AND POPPETS PERFORMED AT VENDOR BY H. S. SOURCE INSPECTION.

CONTAMINATION CONTROL CONTAMINATION CONTROL PROCESSES AND CLEAN AREAS VERIFIED BY INSPECTION. CLEANLINESS VERIFIED BY INSPECTION.

ASSEMBLY/INSTALLATION ASSEMBLY/INSTALLATION REQUIREMENTS VERIFIED BY INSPECTION.

CRITICAL PROCESSES ANODIZE PROCESSING VERIFIED BY INSPECTION. TORQUE OPERATIONS VERIFIED TO H. S. REQUIREMENTS.

TESTING VALVE ACTUATION TEST PERFORMED TO VERIFY FUNCTION. VALVE FUNCTION TESTED AS A DETAIL OF RCRS ASSEMBLY DURING ATP. ATP TESTING VERIFIED BY INSPECTION. VIBRATION TESTING OF ORIGINAL DEVELOPMENT TEST UNIT AS A DETAIL OF RORS ASSEMBLY VERIFIED BY INSPECTION DURING QUALIFICATION. PROOF PRESSURE AND LEAK TEST VERIFIED BY INSPECTION.

HANDLING/PACKAGING HANDLING AND PART PROTECTION MAINTAINED PER H. S. REQUIREMENTS.

- (B) FAILURE HISTORY: NO FAILURE HISTORY.
- (E) OPERATIONAL USE: SHUT DOWN THE RCRS AND INSTALL NEW LIGH CANISTERS FOR COZ REMOVAL. THE LIGH CANISTER SUPPLY IS ADEQUATE FOR 3 DAYS (MINIMUM).

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RELIABILITY MANAGER : T. J. EAVENSON DESIGN ENGINEERING : P. J. CHEN
QUALITY ENGINEERING : E. OCHOA

NASA RELIABILITY MASA SUBSYSTEM MANAGER: NASA QUALITY ASSURANCE :

C. C. Presti in T. Bewood 6/3-/12