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FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL HARDWARE

NUMBER: 06-18-0850-X

SUBSYSTEM NAME: ARS - COOLING

REVISION: 7 06/26/92

PART NAME vendor name PART NUMBER VENDOR NUMBER

a LRU

REGENERABLE CO2 REMOVAL SYSTEM MC623-0016

■ LRU

COMPRESSOR ASSEMBLY

SV806944

s SRU

COMPRESSOR HEAD

SV806923

■ SRÚ : COMPRESSOR MOTOR

SV806924

## PART DATA

■ EXTENDED DESCRIPTION OF PART UNDER ANALYSIS: ULLAGE SAVE COMPRESSOR

- QUANTITY OF LIKE ITEMS: 1
- FUNCTION:

COMPRESSOR REDUCES AMOUNT OF CONSUMABLES LOST BY PUMPING BED PRESSURE FROM 14.7 TO 3.0 PSIA, IN 75 SECONDS, PRIOR TO BED DESORPTION.

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FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL FAILURE MODE

NUMBER: 06-18-0850-01

REVISION# 7 06/26/92 R

SUBSYSTEM: ARS - COOLING

LRU :REGENERABLE CD2 REMOVAL SYSTEM

ITEM NAME: COMPRESSOR MOTOR

CRITICALITY OF THIS FAILURE MODE:2/2

■ FAILURE MODE:

FAILS ON, FAILS OFF

MISSION PHASE:

00

ON-ORBIT

■ VEHICLE/PAYLOAD/KIT EFFECTIVITY: 102 COLUMBIA

: 105 ENDEAVOUR

■ CAUSE:

MECHANICAL SHOCK, VIBRATION, CORROSION, ELECTRICAL SHORT.

■ CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

■ REDUNDANCY SCREEN A) N/A

B) N/A

C) N/A

PASS/FAIL RATIONALE:

■ A)

**=** 8)

= ()

■ MASTER MEAS. LIST NUMBERS: V61P2901A

V61P2902A

V61P2911A

V61P2912A

V61P2922A

- FAILURE EFFECTS -

■ (A) SUBSYSTEM:

FOR FAILS ON. THE CONTROLLER WILL REMOVE POWER FROM THE COMPRESSOR. LOSS OF USE OF THE COMPRESSOR, HOWEVER, THE RCRS WILL CONTINUE TO OPERATE IN THE COMPRESSOR LATCH-OUT MODE.

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- (B) INTERFACING SUBSYSTEM(S): EXCESSIVE USE OF CONSUMABLES AS COMPRESSOR CANNOT BE USED TO RECOVER CABIN AIR FROM SORBENT BED BEFORE EXPOSURE TO VACUUM.
- (C) MISSION:
  POTENTIAL EARLY MISSION TERMINATION DUE TO EXCESSIVE USE OF CONSUMABLES.
- (D) CREW, VEHICLE, AND ELEMENT(S): NO EFFECT.

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■ (E) FUNCTIONAL CRITICALITY EFFECTS:

## - DISPOSITION RATIONALE -

- THE COMPRESSOR IS AN OIL-LESS DESIGN, ROTARY VANE TYPE WITH 4
  CARBON/PHENOLIC VANES, NITRONIC 60 LINER, AND ALUMINUM CYLINDER
  HOUSING. THE MOTOR PROVIDES THE ROTARY POWER TO DRIVE THE ULLAGE
  SAVE COMPRESSOR. THE MOTOR IS AN INDUCTION TYPE, DRIVEN BY 115 VOLT, 3
  PHASE, 400 HZ. 4 WIRE WYE CONNECTED-POWER SUPPLY WITH CASE GROUND. THE
  MOTOR WILL OPERATE WITH A TWO PHASE POWER SUPPLY. HOWEVER, IT WILL NOT
  START UP ON TWO PHASE POWER. THE COMPRESSOR/MOTOR HAS SHOCK MOUNTS.
  MINIMUM OPERATING LIFE OF 4320 HOURS WITH 172,800 START CYCLES FOR 90
  SECONDS EACH ON 11.5 MINUTE INTERVALS.
- # (B) TEST: QUALIFICATION TEST FOR 100 MISSION LIFE: TESTING WITH THE COMPRESSOR INSTALLED AT THE ASSEMBLY LEVEL ONLY. RANDOM VIBRATION INCREASING AT 6 db/oct FROM 20 TO 45 HZ; CONSTANT AT 0.003 g2/HZ FROM 45 TO 1000 HZ; DECREASING AT -6 db/oct FROM 1000 TO 2000 HZ FOR QURATION OF 48 MINUTES PER AXIS IN THREE ORTHOGONAL AXES. SHOCK TEST BY ANALYSIS OF 20 G SAWTOOTH SHOCK IMPULSE FOR 11 MILLISECONDS BURATION.

ACCEPTANCE TEST:

MOTOR AND COMPRESSOR ARE SUBJECTED TO BURN-IN TEST/RUN-IN FOR 16 DAY MISSION SIMULATION TO VERIFY PERFORMANCE.

LIFE/ENDURANCE TEST:

100 MISSIONS EQUIVALENT - 90 SECOND ON CYCLE FOLLOWED BY 11.5 MINUTES COOLING PERIOD.

OMRSD:

ANY TURNAROUND CHECKOUT TESTING IS ACCOMPLISHED IN ACCORDANCE WITH OMRSO AT SYSTEM LEVEL.

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• (C) INSPECTION: RECEIVING INSPECTION INCOMING PARTS/MATERIAL IDENTIFICATION AND CERTIFICATION VERIFIED BY INSPECTION. ANODIZE AND PROTECTIVE FINISH PERFORMED AT VENDOR. RESISTANCE, DIELECTRIC, IR, BURN/RUN IN, SPEED, TORQUE, LOCKED ROTOR AND WEIGHT VERIFICATION PERFORMED AS PART OF VENDOR ATP. VENDOR KIT, SOLDER, PRECAP AND ATP VERIFIED BY H. S. SOURCE INSPECTION.

CONTAMINATION CONTROL
CONTAMINATION CONTROL PROCESSES AND CLEAN AREAS VERIFIED BY INSPECTION.
ASSEMBLY PRECISION CLEAN LEVEL VERIFIED BY INSPECTION.

ASSEMBLY/INSTALLATION
ASSEMBLY AND INSTALLATION OPERATIONS VERIFIED BY INSPECTION. BALANCING VERIFIED BY INSPECTION. CLEARANCE DIMENSIONS VERIFIED BY INSPECTION.

CRITICAL PROCESSES
TORQUE OPERATIONS VERIFIED TO H. S. REQUIREMENTS.

RUN IN, PUMP DOWN, POWER CONSUMPTION, START TIME/CURRENT VERIFIED DURING ASSEMBLY INPROCESS TESTING. FUNCTION VERIFIED DURING RCRS UNIT ATP TESTING WHICH IS VERIFIED BY INSPECTION. VIBRATION TEST OF ORIGINAL DEVELOPMENT TEST UNIT AS A DETAIL OF RCRS ASSEMBLY VERIFIED BY INSPECTION DURING QUALIFICATION.

HANDLING/PACKAGING
HANDLING/PARTS PROTECTION PER H. S. REQUIREMENTS.

- (D) FAILURE HISTORY: NO FAILURE HISTORY.
- (E) OPERATIONAL USE:

  1) SHUT DOWN THE RCRS WHEN CONSUMABLES LEVEL IS LOW.
  - 2) INSTALL NEW LIGH CANISTERS FOR CO2 REMOVAL. THE LIGH CANISTER SUPPLY IS ADEQUATE FOR 3 DAYS.

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FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL FAILURE MODE

NUMBER: 05-18-0850-01

- APPROVALS -

RELIABILITY MANAGER : T. J. EAVENSON DESIGN ENGINEERING : P. J. CHEN QUALITY ENGINEERING : E. OCHOA

NASA RELIABILITY

NASA SUBSYSTEM MANAGER :

NASA QUALITY ASSURANCE :

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