

SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : ACTIVE THERMAL CONTROL FMEA NO 06-3C -0201 -6 REV:08/23
 ASSEMBLY : FREON THERMAL LOOP CRIT. FUNC:
 P/N RI : MC250-0001-0040/0540 CRIT. HDW:
 P/N VENDOR: SV755517
 QUANTITY : 1 VEHICLE 102 103 104
 : ONE PER VEHICLE EFFECTIVITY: X X X
 : PHASE(S): PL LO X OO X DO X LS

PREPARED BY: DES O. TRAN *OT* APPROVED BY: A-PASS B-PASS C-PASS
 REL D. RISING *DR* DES *[Signature]* APPROVED BY (NASA) SSM *[Signature]*
 QE W. SMITH *WS* REL *[Signature]* REL *[Signature]*
 REVISION: 1

ITEM:
INTERCHANGER, WATER/FREON INTERFACE.

FUNCTION:
THE INTERCHANGER TRANSFERS CABIN WASTE HEAT FROM EITHER THE PRIMARY OR SECONDARY WATER COOLANT LOOPS TO THE FREON COOLANT LOOPS.

FAILURE MODE:
RESTRICTED FLOW, WATER.

CAUSE(S):
CORROSION, CONTAMINATION, MECHANICAL SHOCK.

EFFECT(S) ON:
 (A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE
 (A,B) POSSIBLE LOSS OF ONE WATER COOLANT LOOP FOR CABIN COOLING. LOSS COOLING UNTIL IMPLEMENTATION OF CORRECTING ACTION.
 (C) POSSIBLE LOSS OF MISSION. EARLY MISSION TERMINATION FOR LOSS OF ONE WATER COOLANT LOOP.
 (D) SECOND ASSOCIATE FAILURE (LOSS OF REDUNDANT WATER COOLANT LOOP) WILL CAUSE LOSS OF ALL CABIN COOLING AND MAY RESULT IN LOSS OF CREW/VEHICLE.

DISPOSITION & RATIONALE:
 (A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE

(A) DESIGN
 THE INTERCHANGER IS MADE FROM STAINLESS STEEL AND NICKEL BRONZE ALLOYS, WHICH ARE CORROSION RESISTANT AND COMPATIBLE WITH FREON 21 AND WATER. IT CONTAINS NO MOVING PARTS SUBJECT TO WEAR. THE FLOW HEADERS ARE MACHINE-FORMED FROM A SINGLE PIECE STAINLESS STEEL BAR. THE HEADERS ARE WELDED TO THE CORE, WHICH CONTAINS 77 STACKED FIN LAYERS. ALL FINS ARE 0.020 INCHES HIGH AND ARE MADE OF 0.002 INCH THICK STAINLESS STEEL SHEET STOCK. THE FINS ARE RUFFLED AND HAVE A DENSITY OF 32 FLOW PATHS PER INCH. PUMP INLET FILTERS (25 MICRON) PROTECT AGAINST CONTAMINATION.

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(B) TEST

**QUALIFICATION TEST - QUALIFICATION TESTED FOR 100 MISSION LIFE.
VIBRATION TESTED AT 0.075 G²/HZ FOR 52 MIN/AXIS, SHOCK TESTED AT +/- 20
EACH AXIS.**

**ACCEPTANCE TEST - WATER COOLANT LOOPS ARE LEAK CHECKED PRIOR TO EACH
FLIGHT. ATP PRESSURE DROP TEST WILL VERIFY THAT PASSAGES ARE NOT
OBSTRUCTED.**

**OWRSD - WATER IS SERVICED THROUGH A 10 MICRON GSE FILTER AND IS ANALYZE
PER SE-S-0073**

(C) INSPECTION

RECEIVING INSPECTION

**RAW MATERIAL AND PURCHASED COMPONENTS REQUIREMENTS ARE VERIFIED BY
INSPECTION. PARTS PROTECTION IS VERIFIED BY INSPECTION.**

CONTAMINATION CONTROL

**SYSTEMS FLUID ANALYSES FOR CONTAMINATION ARE VERIFIED BY INSPECTION.
CONTAMINATION CONTROL PLAN IS VERIFIED BY INSPECTION. CONTAMINATION
CONTROL PROCESSES AND CLEAN AREAS ARE VERIFIED BY INSPECTION.**

ASSEMBLY/INSTALLATION

**MANUFACTURING, INSTALLATION, AND ASSEMBLY OPERATIONS ARE VERIFIED BY
INSPECTION. SHEET METAL PARTS ARE INSPECTED AND VERIFIED BY INSPECTION
SURFACE FINISHES VERIFIED BY INSPECTION. DIMENSIONS VERIFIED BY
INSPECTION.**

CRITICAL PROCESSES

**WELDING IS VERIFIED BY INSPECTION. ALL WELDS ARE STRESS RELIEVED AFTER
WELDING, VERIFIED BY INSPECTION. BRAZING IS VERIFIED BY INSPECTION.**

NONDESTRUCTIVE EVALUATION

**HEADER WELDS TO THE TUBES ARE PENETRANT AND X-RAY INSPECTED. OTHER
WELDS (MOUNTING PADS AND HEADER WELDS TO THE CORES) ARE PENETRANT AND
10X MAGNIFICATION VISUALLY INSPECTED. BRAZES ARE VERIFIED BY PROOF AND
LEAK TESTS.**

TESTING

**INSPECTION VERIFIES THAT RESULTS OF ACCEPTANCE TESTING AND FLOWRATES AR
WITHIN SPECIFIED LIMITS.**

HANDLING/PACKAGING

HANDLING AND PACKAGING REQUIREMENTS VERIFIED BY INSPECTION.

(D) FAILURE HISTORY

NO FAILURE HISTORY.

(E) OPERATIONAL USE

**ON-BOARD ALARM, WATER PUMP DELTA PRESSURE, WILL PROVIDE INDICATION OF
HARDWARE FAILURE. ACTIVATE REDONDANT WATER LOOP PUMP. ENTRY AT NEXT
PLS.**