PRINT DATE: 02/19/98

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FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CIL HARDWARE NUMBER: 06-3D-0501 -X

SUBSYSTEM NAME: ATCS - RADIATORS AND FLOW CONTROL

REVISION: 1 12/02/97

PART DATA		
	PART NAME	PART NUMBER
	VENDOR NAME	VENDOR NUMBER
	: RADIATOR AND FLOW CONTROL ASSY	MC203-0002-XXXX
		224-00XX-XXX
J :	: FWD RADIATOR RH	MC203-0002-0012
		224-00010-109
	: MID-FWD RADIATOR RH	MC203-0002-0019
		224-00015-113
	: MID-AFT RADIATOR RH	MC203-0002-0022
		224-00020-109
	: AFT RADIATOR RH	MC203-0002-0028
		224-00025-109
	: FWD RADIATOR LH	MC203-0002-0032
		224-00030-109
₹U	: MID-FWD RADIATOR LH	MCZ03-0002-0039
J		224-00035-113
	: MID-AFT RADIATOR LH	MC203-0002-0041
		224-00040-109
	: AFT RADIATOR LH	MC203-0002-0048
		224-00045-109

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:

SPACE RADIATOR PANEL ASSEMBLY.

REFERENCE DESIGNATORS:

QUANTITY OF LIKE ITEMS: 8

FOUR PER LOOP; EIGHT PER VEHICLE

FUNCTION:

PROVIDES HEAT REJECTION SURFACES FOR FREON COOLANT LOOPS WITH RADIATOR PANELS DEPLOYED. USED DURING ORBITAL OPERATIONS. FOUR PANELS ON LOOP 1 (LEFT SIDE OF VEHICLE) AND FOUR PANELS ON LOOP 2 (RIGHT SIDE OF VEHICLE).

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FAILURE MODES EFFECTS ANALYSIS (FMEA) --CIL HARDWARE NUMBER: 06-3D-0501-X

NORMAL ATCS ENTRY, THE RADIATORS ARE COLD SOAKED PRIOR TO DOOR CLOSURE AND ARE USED FOR THERMAL CONTROL AFTER FLASH EVAPORATOR GPC SHUTDOWN.

SHUTTLE CRITICAL ITEMS LIST - ORBITER

FMEA NO 06-3D -0501 -3 REV:08/29/82 SUBSYSTEM : ACTIVE THERMAL CONTROL

ASSEMBLY : RADIATOR & FLOW CONTROL

CRIT. FUNC:

:MC203-0002-XXXX P'N RI

CRIT. HDW:

P/N VENDOR: 224-000XX-XXX

103 104

8: YTITMAUO

102 VEHICLE Х X EFFECTIVITY:

:EIGHT/VEHICLE : FOUR PER LOOP

LO 00 X D0 PHASE(5): PL

REDUNDANCY SCREEN:

A-PASS B-PASS C-PASS

PREPARED BY:

APPROVED BY

APPROVED BY (NASA)

DES REL

O. TRAN AT DES D. RISING DW-REL SSM Il R man REL

QΣ

w. smith Aloe

ITEM:

SPACE RADIATOR PANEL ASSEMBLY.

FUNCTION:

PROVIDES HEAT REJECTION SURFACES FOR FREON COOLANT LOOPS WITH RADIATOR PANELS DEPLOYED. USED DURING ORBITAL OPERATIONS. FOUR PANELS ON LOOP 1 (LEFT SIDE OF VEHICLE) AND FOUR PANELS ON LOOP 2 (RIGHT SIDE OF VEHICLE). NORMAL ATCS ENTRY, THE RADIATORS ARE COLD SOAKED PRIOR TO DOCK CLOSURE AND ARE USED FOR THERMAL CONTROL AFTER FLASH EVAPORATOR GPC SHUTDOWN.

ROCKWELL PART NUMBER	VENDOR PART NUMBER
MC203-0002-0012	224-00010-109
MC203-0002-0019	224-00015-113
MC203-0002-0022	224-00020-109
MC203-0002-0028	224-00025-109
MCZ03-0002-0032	224-00030-109
MC203-0002-0039	224-00035-113
MC203-0002-0041	224-00040-109
MC203-0002-0048	224-00045-109
	MC203-0002-0012 MC203-0002-0019 MC203-0002-0022 MC203-0002-0028 MC203-0002-0032 MC203-0002-0039 MC203-0002-0041

PATILIRE MODE:

RESTRICTED FLOW OF FREON 21 IN PANEL ASSEMBLY

CAUSE(S):

CONTAMINATION, MECHANICAL SHOCK.

EFFECT(S) ON:

- (A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE
- (A,B) POSSIBLE LOSS OF RADIATOR FLOW IN ONE FREON LOOP FOR VEHICLE COOLING THROUGH THE RADIATORS.
- (C) POSSIBLE LOSS OF MISSION DUE TO THE LOSS RADIATOR COOLING TO SUPPORT PAYLOAD OPERATIONS. A CRITICALITY 2/2 CONDITION.
- (D) NO EFFECT.

SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : ACTIVE THERMAL CONTROL FMEA NO 06-30 -0501 -3 REV:08/29/8

(E) I NOTIONAL CRITICALITY EFFECT - ANY TWO ADDITIONAL FAILURES (OTHER FREON COOLANT LOOP, HI-LOAD EVAPORATOR, AND AMMONIA BOILER SYSTEM) WILL CAUSE LOSS OF VEHICLE COOLING CAPABILITY AND MAY RESULT IN LOSS OF CREW/VEHICLE, A CRITICALITY LR3 (PPP) CONDITION.

DISPOSITION & RATIONALE:

(A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE

(A) DESIGN

DESIGN SAFETY FEATURE OF MULTI-PARALLEL PASSAGES. INSIDE DIAMETER OF THE TUBES ARE 0.187 AND 0.236 INCHES FOR THE FORWARD AND AFT PANELS RESPECTIVELY. TUBES ARE 6061-T6 ALUMINUM WHICH IS COMPATIBLE WITH FREON 21. RADIATOR PANELS AND INTERPANEL PLUMBING ARE DESIGNED FOR 400 MISSIC LIFE. FILTRATION PROVISIONS MINIMIZE CONTAMINATION (65 MICRON FILTERS I THE FLOW CONTROL ASSEMBLY, 25 AND 61 MICRON FILTERS IN FUMP PACKAGE).

(B) TEST

QUALIFICATION TEST - THE RADIATOR PANELS AND INTER-PANEL PLUMBING WAS LIFE TESTED FOR AN EQUIVALENT OF 400 MISSIONS (FOUR LIFETIMES). VIBRATION TESTED AT 3 G^2/HZ FOR Y AND Z AXES, AND $1G^2/HZ$ FOR X AXIS, FOR 48 MIN/AXIS. SHOCK TESTED AT +/- 20 G EACH AXIS.

ACCEPTANCE TEST - ATP VERIFIES UNRESTRICTED FLOW.

OMPSD - FCL FLOW RATES ARE CHECKED FRIOR TO EACH FLIGHT. FREON CHEMICAL ANALYSIS PER SI-S-0073 DURING SERVICING. FREON IS SERVICED THROUGH A FINAL FILTER OF 25 MICRON SIZE.

(C) INSPECTION

RECEIVING INSPECTION

RAW MATERIAL IS VERIFIED BY INSPECTION. VISUAL INSPECTION/ID PERFORMED. PARTS PROTECTION IS VERIFIED BY INSPECTION.

CONTAMINATION CONTROL

SYSTEM FLUID SAMPLES PERIODICALLY ANALYZED FOR CONTAMINATION AND VERIFIE BY INSPECTION. CONTAMINATION CONTROL PROCESSES AND CORROSION PROTECTION PROVISIONS ARE VERIFIED BY INSPECTION. FORMAL CONTAMINATION CONTROL PLA IS VERIFIED BY INSPECTION. INSPECTION VERIFIES CLEANLINESS TO LEVEL 200

ASSEMBLY/INSTALLATION

MANUFACTURING, INSTALLATION AND ASSEMBLY OPERATIONS ARE VERIFIED BY INSPECTION ON SHOP TRAVELER MIPS. MATERIAL AND EQUIPMENT CONFORMANCE TO CONTRACT REQUIREMENTS ARE VERIFIED BY INSPECTION. PROCESSING EQUIPMENT CONTROLS ARE VERIFIED BY INSPECTION.

CRITICAL PROCESSES

WELDING IS VERIFIED BY INSPECTION.

NONDESTRUCTIVE EVALUATION

X-RAY EXAMINATION OF FUSION WELDS IS VERIFIED BY INSPECTION.

SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : ACTIVE THERMAL CONTROL FMEA NO 06-3D -0501 -3 REV:08/29/5

TESTING
FUNCTIONAL TEST IS MONITORED BY INSPECTION TO VERIFY FLOWRATE IS WITHIN SPECIFIED LIMITS.

HANDLING/PACKAGING PROPERLY MONITORED HANDLING AND STORAGE ENVIRONMENTS ARE VERIFIED BY INSPECTION.

- (D) PAILURE HISTORY NO FAILURE HISTORY.
- (E) OPERATIONAL USE
 ON-BOARD ALARM, EVAP OUT TEMPERATURE, WILL PROVIDE INDICATION OF HARDWAF
 FAILURE. LOSS OF RADIATOR MAY CAUSE AN EARLY END OF MISSION. RADIATOR
 WILL BE BYPASSED, FREON PUMP WILL BE TURNED OFF, AND A POWERDOWN WILL BE
 PERFORMED. FREON PUMP WILL BE REACTIVATED FOR ENTRY.