

PRINT DATE: 09/09/92

FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL HARDWARE
 NUMBER: M5-6MB-2077-G-X

SUBSYSTEM NAME: ELECTRICAL POWER GENERATION - CRYO, GENERIC
 REVISION : 9 09/09/92

	PART NAME VENDOR NAME	PART NUMBER VENDOR NUMBER
LRU	: PANEL R1A2	VO70-730276
SRU	: RESISTOR	RWR80S1211FR

- PART DATA -

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:

RESISTOR, CURRENT LIMIT, 1.2K OHM, 2 WATT - 02 MANIFOLD VALVE, TANKS 1 AND 2

REFERENCE DESIGNATORS: 32V73A1A2A1R2
 : 32V73A1A2A1R6
 : 32V73A1A2A1R10
 : 32V73A1A2A1R16

QUANTITY OF LIKE ITEMS: 4
 FOUR, TWO PER 02 MANIFOLD VALVE CIRCUIT

FUNCTION:

PROVIDES CURRENT LIMIT/CIRCUIT PROTECTION FOR THE CONTROL CIRCUIT OF 02
 MANIFOLD ISOLATION VALVES 40V45LV011 AND 40V45LV021.

FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL FAILURE MODE
NUMBER: M5-6MB-2077-G-01

REVISION# 9 09/09/92

SUBSYSTEM: ELECTRICAL POWER GENERATION - CRYO, GENERIC
LRU PANEL RL2Z
ITEM NAME: RESISTOR
CRITICALITY OF THIS FAILURE MODE:1R2

FAILURE MODE:
OPEN

MISSION PHASE:
LO LIFT-OFF
OO ON-ORBIT
DO DE-ORBIT

VEHICLE/PAYLOAD/KIT EFFECTIVITY: 102 COLUMBIA
: 103 DISCOVERY
: 104 ATLANTIS
: 105 ENDEAVOUR

CAUSE:
STRUCTURAL FAILURE (MECHANICAL STRESS, VIBRATION), ELECTRICAL STRESS, THERMAL STRESS, PROCESSING ANOMALY

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN A) PASS
B) FAIL
C) PASS

PASS/FAIL RATIONALE:
A)

B)
REDUNDANCY SCREEN "B" FAILS EVEN THOUGH THE FAILURE OF THIS RESISTOR IS DETECTABLE BECAUSE THE TIME FOR CORRECTIVE ACTION (ELECTRICAL LOAD RECONFIGURATION) EXCEEDS THE TIME TO EFFECT (MANIFOLD GROSS EXTERNAL LEAK STARVES TWO FCP'S DURING ASCENT/DESCENT).

C)

- FAILURE EFFECTS -

(A) SUBSYSTEM:
LOSS OF ABILITY TO OPEN AFFECTED MANIFOLD VALVE AFTER INADVERTENT OR COMMANDED VALVE CLOSURE.

FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL FAILURE MODE
 NUMBER: M5-6MB-2077-G-01

(B) INTERFACING SUBSYSTEM(S):
 SAME AS (A)

(C) MISSION:
 (CRIT 2/2) POSSIBLE LOSS OF MISSION DUE TO ASSOCIATED MANIFOLD VALVE FAILING CLOSED RESULTING IN ONE TANK BEING ISOLATED TO A SINGLE FUEL CELL. MISSION TERMINATED WHEN THE OXYGEN IN THAT TANK IS CONSUMED.

(D) CREW, VEHICLE, AND ELEMENT(S):
 FIRST FAILURE - NO EFFECT

(E) FUNCTIONAL CRITICALITY EFFECTS:
 (CRIT 1R2) POSSIBLE LOSS OF CREW/VEHICLE DUE TO THE FOLLOWING SCENARIO:
 1) RESISTOR OPENS (VALVE REMAINS OPEN), AND 2) GROSS EXTERNAL LEAK STARVES TWO FCP'S (LOSS OF TWO FCP'S DURING ASCENT LOSES CREW/VEHICLE. LOSS OF A SECOND FCP DURING DESCENT LOSES CREW/VEHICLE IF INSUFFICIENT TIME IS AVAILABLE FOR AN ELECTRICAL LOAD RECONFIGURATION RESULTING IN THE INABILITY OF THE SINGLE REMAINING FUEL CELL TO SUPPLY ADEQUATE ELECTRICAL POWER.)

 - DISPOSITION RATIONALE -

(A) DESIGN:
 REFER TO APPENDIX E, ITEM NO. 3 - RESISTOR

(B) TEST:
 REFER TO APPENDIX E, ITEM NO. 3 - RESISTOR

RESISTOR INTEGRITY IS FUNCTIONALLY VERIFIED DURING FLIGHT. PERFORM GROUND TURNAROUND TEST WHEN VALID VERIFICATION IS UNOBTAINABLE IN FLIGHT OR AFTER LRU REPLACEMENT.

(C) INSPECTION:
 REFER TO APPENDIX E, ITEM NO. 3 - RESISTOR

(D) FAILURE HISTORY:
 REFER TO APPENDIX E, ITEM NO. 3 - RESISTOR

(E) OPERATIONAL USE:
 NO CREW ACTION AFTER FIRST FAILURE.

FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL FAILURE MODE

NUMBER: M5-6MB-2077-G-01

- APPROVALS -

PRODUCT ASSURANCE MGR : T. J. EAVENSON
 PRODUCT ASSURANCE ENG : T. K. KIMURA
 DESIGN ENG TEAM LEADER : G. M. ANDERSON
 DESIGN ENGINEERING : T. D. NGUYEN
 NASA RELIABILITY :
 NASA SUBSYSTEM MANAGER :
 NASA EPD&C RELIABILITY :
 NASA QUALITY ASSURANCE :
 NASA EPD&C SUBSYS MGR :

T. J. Eavenson 9/15/92
J. K. Kimura 9/14/92
G. M. Anderson 9/15/92
T. D. Nguyen 9/15/92
197765 J. H. Steadinger 12/16/92
197765 G. M. Anderson 12/16/92
Daniel Lopez for S. Woodard 10/14/92
MTC KO [Signature] 11/9/92
[Signature] for F. Abair 14 Dec 92