

**SHUTTLE CRITICAL ITEMS LIST - ORBITER**

SUBSYSTEM : P/L RETEN & DEPLOY-MPM DEPLOY FMEA NO 02-5B-P06-1 REV:04/05/88

ASSEMBLY :MPM DEPLOYMENT MECHANISM			CRIT. FUNC: 1R
P/N RI :V082-544600			CRIT. HDW: 2
P/N VENDOR:	VEHICLE	102	103 104
QUANTITY :1	EFFECTIVITY:	X	X X
	PHASE(S):	PL LO	OO X DO X LS

	REDUNDANCY SCREEN: A-PASS B-PASS C-PASS		
PREPARED BY:	APPROVED BY: <i>A. Sampson</i>	APPROVED BY: (NASA):	
DES D. S. CHEUNG	DES <i>DEC For G. CAMPBELL</i>	SSM	
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ITEM:  
DRIVE LINKAGE, SHOULDER

FUNCTION:  
REDUNDANT POWER DRIVE UNIT (PDU) MOTORS DRIVE THROUGH TORQUE LIMITERS AND THE PDU GEARBOX TO PROVIDE TORQUE TO THE MANIPULATOR POSITIONING MECHANISM (MPM) DRIVESHAFT WHICH IN TURN DRIVES THE SHOULDER AND FORWARD/MID/AFT PEDESTAL ROTARY DRIVE GEARBOX/DRIVE LINKAGES. THE SHOULDER DRIVE LINKAGES ALSO INCLUDE THE SHOULDER HOOK WHICH IS ENGAGED DURING DEPLOYMENT TO ESTABLISH THE PRIMARY LOAD BEARING PATH.

FAILURE MODE:  
PHYSICAL BINDING/JAMMING

CAUSE(S):  
ADVERSE TOLERANCES/WEAR, CONTAMINATION/FOREIGN OBJECT/DEBRIS, DEFECTIVE PART/MATERIAL OR MANUFACTURING DEFECT, TEMPERATURE, VIBRATION

EFFECTS ON:  
(A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE

(A) FAILURE WILL RESULT IN LOSS OF ABILITY TO POSITION THE SHOULDER. TORQUE WILL FEED BACK INTO THE SYSTEM AND SLIP THE TORQUE LIMITERS IN THE PDU AND STALL POSITIONING OF ALL MPM.

(B) FAILURE WILL RESULT IN LOSS OF ABILITY TO POSITION MPM CAUSING POTENTIAL INTERFERENCE WITH PAYLOAD BAY (PLB) DOOR CLOSURE.

(C) FAILURE WILL RESULT IN POSSIBLE LOSS OF MISSION DUE TO BLOCKAGE OF PAYLOAD DEPLOYMENT/RETRIEVAL ENVELOPE OR INABILITY TO ESTABLISH LOAD PATH FOR REMOTE MANIPULATOR SYSTEM (RMS).

(D) FAILURE WILL REQUIRE JETTISON OF MPM TO PREVENT POSSIBLE LOSS OF CREW/VEHICLE DUE TO INTERFERENCE WITH PLB DOOR CLOSURE.

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DISPOSITION & RATIONALE:

(A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE

(A) DESIGN

DUAL ROTATING SURFACES PROVIDED AT ALL JOINTS. LATCH MECHANISM POSITIVELY DRIVEN BY SAME ACTUATOR WHICH PROVIDES POSITIONING FORCES. LINKAGE REQUIRED TO EXHIBIT STRUCTURAL FACTORS EXCESS OF 1.4. MECHANISM RIGGED AND POST RIGGING INSPECTIONS CONTROLLED BY ROCKWELL SPECIFICATION ML0308-0126.

(B) TEST

QUALIFICATION TESTS: THE MPM DEPLOYMENT ACTUATOR MC287-0037-0006/-0007 IS CERTIFIED PER CR-29-287-0037-0001G (REF FMEA/CIL 02-5B-P01-3) THE MANIPULATOR POSITIONING MECHANISM INSTALLATION IS CERTIFIED PER CR-44-000002E. THE SYSTEM INSTALLATION QUALIFICATION TEST INCLUDED: ACCEPTANCE (TO CONFIRM ALL COMPONENTS HAVE BEEN ASSEMBLED AND RIGGED PER APPLICABLE DRAWINGS AND SPECIFICATIONS); FLIGHT VIBRATION - 20 TO 2,000 HZ RANGE WITH MAXIMUM OF 0.006 g<sup>2</sup>/HZ FROM 100 TO 250 HZ FOR 49.5 MINS/AXIS AT LEVEL "A", AND WITH MAXIMUM OF 0.047 g<sup>2</sup>/HZ FROM 50 TO 250 HZ FOR 49.5 MINS/AXIS AT LEVEL "B"; STIFFNESS TEST - APPLIED LOADS AND MOMENTS (11 CONDITIONS) TO THE SHOULDER MECHANISM (8 CONDITIONS) AND RETENTION FITTING (3 CONDITIONS); LIMIT LOAD - APPLIED LIMIT LOAD AND 115% OF LIMIT LOAD TO THE RETENTION FITTING AND SHOULDER MECHANISM (STOWED AND DEPLOYED POSITIONS); FUNCTIONAL CHECKOUT WITHOUT MANIPULATOR ARM - CYCLED MPM WITH BOTH MOTORS, 40 SEC MAX/DEPLOY STROKE AND 50 SEC MAX/STOWED STROKE; FUNCTIONAL CHECKOUT WITH MANIPULATOR ARM - CYCLED EACH RETENTION LATCH TO THE LATCHED AND UNLATCHED POSITION WITH BOTH MOTORS, 7.5 SEC MAX/LATCH AND UNLATCH STROKE AND REPEATED DEPLOY AND STOW CYCLES OF MPM.

QUAL TESTS ALSO INCLUDE: HORIZONTAL OPERATION - CYCLED 115 TIMES AT +70 DEG F, 60 TIMES AT +25 DEG F, 100 TIMES AT +168 DEG F WITH ENGINEERING ARM INSTALLED CYCLED 100 TIMES AT -100 DEG F AND 100 TIMES AT +250 DEG F WITHOUT THE ENGINEERING ARM INSTALLED; SEPARATION SHOULDER/PEDESTAL - PERFORMED 4 PYRO SEPARATIONS (2 FOR SHOULDER AND 2 FOR RETENTION FITTING); READY-TO-LATCH INDICATION - OPERATED STRIKER BAR 500 TIMES AT AMBIENT TEMPERATURE, 20 TIMES AT -50 DEG F, 500 TIMES AT -100 DEG F AND 500 TIMES AT +168 DEG F; LIMIT LOAD (LANDING CASE) - APPLIED LIMIT LOADS AND 115% LIMIT LOADS TO SHOULDER MECHANISM IN STOWED POSITION; MECHANICAL STOP TEST - THE MPM DRIVE MECHANISM WAS OPERATED INTO ITS STOPS TEN TIMES; DELTA QUAL TEST - WITH DOWEL PIN INSTALLED THE SHOULDER MECHANISM IN DEPLOYED POSITION WAS SUBJECTED TO LIMIT LOADS; VERTICAL OPERATIONS - CONDUCTED 75 CYCLES AT ROOM AMBIENT CONDITIONS; ULTIMATE LOADS - CONDUCTED ULTIMATE LOADS ON RETENTION FITTING AND ON SHOULDER MECHANISM; PYRO SEPARATION - WITH DOWEL PIN INITIATED PYRO SEPARATION.

ACCEPTANCE TESTS: THE MPM ACCEPTANCE TEST CONSISTED OF CONFIRMATION OF ACCEPTANCE DATA APPLICABLE TO ASSEMBLY AND RIGGING.

OMRSD: GROUND TURNAROUND INCLUDES PORT MPM DEPLOY (SYSTEM 1), PORT MPM STOW (SYSTEM 1), PORT MPM DEPLOY (SYSTEM 2), AND PORT MPM STOW (SYSTEM 2).

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(C) INSPECTION

RECEIVING INSPECTION

MATERIAL AND PROCESS CERTIFICATIONS ARE VERIFIED BY RECEIVING INSPECTION.

CONTAMINATION CONTROL

CORROSION PROTECTION IS REQUIRED AND VERIFIED BY INSPECTION. CLEANLINESS IS MAINTAINED PER APPLICABLE SPECIFICATION AND VERIFIED BY INSPECTION.

ASSEMBLY/INSTALLATION

ALL DETAILS AND ASSEMBLY COMPONENTS ARE MADE TO DRAWING REQUIREMENTS AND APPLICABLE SPECIFICATIONS. ALL ARE VERIFIED BY INSPECTION ON MANUFACTURING ORDERS. INSTALLATION OF THREADED FASTENERS VERIFIED BY INSPECTION. PLANNED SEQUENCES TO ACHIEVE RIGGING CONFIGURATION ARE VERIFIED BY INSPECTION, INCLUDING TORQUING VERIFICATION AND SAFETY WIRING PER DRAWING. ELECTRICAL OPERATION IS PERFORMED BY USING GSE GROUND OPERATION CHECKOUT CONSOLE C70-0863, OR EQUIVALENT TEST; MANUFACTURING ORDER IS REQUIRED AND VERIFIED BY INSPECTION. MPM MECHANISM ASSEMBLY CONFIGURATION IS RIGGED AND POST-RIGGING INSPECTION CONTROLLED BY ROCKWELL.

NONDESTRUCTIVE EVALUATION

PENETRANT INSPECTION IS VERIFIED BY INSPECTION.

CRITICAL PROCESSES

ELECTRICAL BOND AND TEST ARE VERIFIED BY INSPECTION. APPLICATION OF DRY FILM LUBE IS VERIFIED BY INSPECTION.

TESTING

ATP IS OBSERVED AND VERIFIED PER PROCEDURE.

HANDLING/PACKAGING

PARTS ARE PACKAGED PER APPLICABLE SPECIFICATION AND VERIFIED BY INSPECTION.

(D) FAILURE HISTORY

THERE HAVE BEEN NO ACCEPTANCE TEST, QUALIFICATION TEST, FIELD OR FLIGHT FAILURES ASSOCIATED WITH THIS FAILURE MODE.

(E) OPERATIONAL USE

THE MPM MAY BE JETTISONED IF PREVENTING PLB DOOR CLOSURE.