

**FAILURE MODES EFFECTS ANALYSIS (FMEA) -- NON-CIL HARDWARE  
NUMBER: 05-6-2188 -X**

**SUBSYSTEM NAME: ELECTRICAL POWER DISTRIBUTION & CONTROL**  
**REVISION: 1** 07/26/99

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**PART DATA**

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	<b>PART NAME</b>	<b>PART NUMBER</b>
	<b>VENDOR NAME</b>	<b>VENDOR NUMBER</b>
LRU	: MDCA 1	V070-764200
LRU	: MDCA 2	V070-764220
LRU	: MDCA 3	V070-764230
SRU	: DIODE	JANTX1N1188R

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**EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:**  
DIODE, ISOLATION, 35 AMP - MAIN DC BUS TO ESSENTIAL BUS ISOLATION

**REFERENCE DESIGNATORS:** 40V76A31CR2  
40V76A31CR3  
40V76A32CR2  
40V76A32CR3  
40V76A33CR2  
40V76A33CR3

**QUANTITY OF LIKE ITEMS: 6**  
SIX REQUIRED - ONE PER MAIN BUS POWER CIRCUIT

**FUNCTION:**  
ISOLATES EACH MAIN DC BUS FEEDER TO THE ESSENTIAL BUS FROM THE OTHER  
POWER SOURCES OF THE ESSENTIAL BUS.

## FAILURE MODES EFFECTS ANALYSIS FMEA -- NON-CIL FAILURE MODE

NUMBER: 05-6-2186-02

REVISION#: 1 07/26/99

SUBSYSTEM NAME: ELECTRICAL POWER DISTRIBUTION &amp; CONTROL

LRU: MDCA 1, 2, 3

ITEM NAME: DIODE

CRITICALITY OF THIS

FAILURE MODE: 1R3

## FAILURE MODE:

SHORT (END TO END)

## MISSION PHASE:

PL	PRE-LAUNCH
LO	LIFT-OFF
OO	ON-ORBIT
DO	DE-ORBIT
LS	LANDING/SAFING

## VEHICLE/PAYLOAD/KIT EFFECTIVITY:

102	COLUMBIA
103	DISCOVERY
104	ATLANTIS
105	ENDEAVOUR

## CAUSE:

THERMAL STRESS, CONTAMINATION, STRUCTURAL FAILURE (MECHANICAL STRESS, VIBRATION), ELECTRICAL STRESS, PROCESSING ANOMALY

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

## REDUNDANCY SCREEN

A) PASS  
B) N/A  
C) PASS

## PASS/FAIL RATIONALE:

A)

B)

"B" SCREEN IS "N/A" BECAUSE FAILURE OF AT LEAST TWO REMAINING PATHS IS READILY DETECTABLE DURING FLIGHT (DIODE SHORT TO STRUCTURE, POWER CONTACTOR, REDUNDANT REACTANT VALVE CLOSURE).

C)

## - FAILURE EFFECTS -

(A) SUBSYSTEM:

**FAILURE MODES EFFECTS ANALYSIS (FMEA) -- NON-CIL FAILURE MODE**  
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FIRST FAILURE - NO EFFECT LOSS OF NORMAL ISOLATION

**(B) INTERFACING SUBSYSTEM(S):**  
 SAME AS (A)

**(C) MISSION:**  
 NO EFFECT - FIRST FAILURE

**(D) CREW, VEHICLE, AND ELEMENT(S):**  
 NO EFFECT - FIRST FAILURE

**(E) FUNCTIONAL CRITICALITY EFFECTS:**

POSSIBLE LOSS OF CREW/VEHICLE AFTER FOURTH FAILURE (ASSOCIATED FUEL CELL TO MAIN DC BUS POWER CONTACTOR FAILED CLOSED) DUE TO INABILITY TO "SAFE" A FUEL CELL. LOSS OF AN ESSENTIAL BUS (REQUIRES TWO FAILURES ON SAME DIODE - SHORT AND SHORT TO GROUND ON DIODE CR2 OR CR3) RESULTS IN LOSS OF THE ASSOCIATED FUEL CELL COOLANT PUMP AS WELL AS REDUNDANT CONTROL OF THAT FUEL CELL'S REACTANT VALVES. THIS NECESSITATES REMOVAL OF ALL LOAD FROM THE FUEL CELL IN ORDER TO RENDER IT SAFE. INABILITY TO REDUNDANTLY CLOSE REACTANT VALVES OR REMOVE THE BUS LOAD FROM THE FUEL CELL UNDER THESE CIRCUMSTANCES, WILL RESULT IN FUEL CELL OVERHEATING WITH SUBSEQUENT RUPTURE AND/OR EXPLOSION/FIRE.

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- APPROVALS -

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EDITORIALLY APPROVED  
 TECHNICAL APPROVAL

: BNA  
 : VIA APPROVAL FORM

: J. Kamura 7-26-99  
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