

FAILURE MODES EFFECTS ANALYSIS (FMEA) -- NON-CIL HARDWARE
NUMBER: 05-6-2230 -X

SUBSYSTEM NAME: ELECTRICAL POWER DISTRIBUTION & CONTROL
REVISION: 1 07/26/99

PART DATA

| | PART NAME | PART NUMBER |
|-----|--------------------|----------------------|
| | VENDOR NAME | VENDOR NUMBER |
| LRU | : PANEL R1A1 | V070-730275 |
| SRU | : SWITCH, TOGGLE | ME452-0102-7102 |

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:
SWITCH, TOGGLE, SPDT, "ON"/"OFF" (MOMENTARY) - CONTROL BUS POWER INPUT/
RESET

REFERENCE DESIGNATORS: 32V73A1A1S1
32V73A1A1S2
32V73A1A1S3

QUANTITY OF LIKE ITEMS: 3
THREE

FUNCTION:
PROVIDES MANUAL RESET CONTROL OF CONTROL BUS REMOTE POWER
CONTROLLERS.

FAILURE MODES EFFECTS ANALYSIS FMEA – NON-CIL FAILURE MODE

NUMBER: 05-6-2230-02

REVISION#: 1 07/26/99

SUBSYSTEM NAME: ELECTRICAL POWER DISTRIBUTION & CONTROL

LRU: PANEL R1A1

CRITICALITY OF THIS

ITEM NAME: SWITCH, TOGGLE

FAILURE MODE: 1R3

FAILURE MODE:

FAILS CLOSED

MISSION PHASE:

LO LIFT-OFF
 OO ON-ORBIT
 DO DE-ORBIT
 LS LANDING/SAFING

VEHICLE/PAYLOAD/KIT EFFECTIVITY:

| | |
|-----|-----------|
| 102 | COLUMBIA |
| 103 | DISCOVERY |
| 104 | ATLANTIS |
| 105 | ENDEAVOUR |

CAUSE:

PIECE PART STRUCTURAL FAILURE, CONTAMINATION, VIBRATION, MECHANICAL SHOCK,
 PROCESSING ANOMALY

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN

| |
|---------|
| A) PASS |
| B) N/A |
| C) PASS |

PASS/FAIL RATIONALE:

A)

B)

"B" SCREEN IS "N/A" BECAUSE FAILURE OF AT LEAST TWO REMAINING PATHS IS READILY
 DETECTABLE IN FLIGHT.

C)

- FAILURE EFFECTS -**(A) SUBSYSTEM:**

LOSS OF CAPABILITY TO RESET AN INADVERTENTLY TRIPPED RPC.

**FAILURE MODES EFFECTS ANALYSIS (FMEA) -- NON-CIL FAILURE MODE
NUMBER: 05-6-2230- 02**

(B) INTERFACING SUBSYSTEM(S):
NO EFFECT - FIRST FAILURE

(C) MISSION:
NO EFFECT - FIRST FAILURE

(D) CREW, VEHICLE, AND ELEMENT(S):
NO EFFECT - FIRST FAILURE

(E) FUNCTIONAL CRITICALITY EFFECTS:
POSSIBLE LOSS OF CREW/VEHICLE DUE TO THE FOLLOWING SCENARIO:

1. SWITCH FAILS CLOSED
2. INADVERTENT RPC TRIP - LOSS OF ONE OF THREE POWER SOURCES TO TWO CONTROL BUSES DUE TO THE INABILITY TO RESET.
3. REDUNDANT RPC FAILS "OFF" - LOSS OF SECOND POWER SOURCE TO TWO CONTROL BUSES.
4. FUSE IN REDUNDANT PATH FAILS OPEN - LOSS OF ONE CONTROL BUS.
5. LOSS OF A SECOND CONTROL BUS.

RESULTS IN LOSS OF CONTROL POWER REQUIRED FOR OPERATIONS OF CRITICAL FUNCTIONS.

- APPROVALS -

| | | |
|----------------------|---------------------|----------------------------|
| EDITORIALLY APPROVED | : BNA | : <u>J. Kemura 7-26-99</u> |
| TECHNICAL APPROVAL | : VIA APPROVAL FORM | : 96-CIL-025_05-6 |