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PRINT DATE: 07/23/90

FAILURE MODES EFFECTS ANALYSIS (FNEA) - CRITICAL HARDWARE

NUMBER: 05-6AB-2126-X

SUBSYSTEM NAME: EPD&C - ACTUATOR, VENT DOORS

REVISION: 2 07/20/90

		PART NAME VENDOR NAME	PART MUMBER VENDOR NUMBER	
LRU	:	FWD MCA-I	V070-763600	
LŔŨ	:	FWD MCA-2	V070-763620	
LRU	:	FWD MCA-3	Y070-753630	
ŲЯZ	:	RELAY, HYBRID	MC455-0135-0001	
		PART DATA	**	

■ EXTENDED DESCRIPTION OF PART UNDER ANALYSIS: RELAY, HYBRID, FOUR POLE, NON-LATCH, RIGHT AND LEFT FORWARD VENT DOORS 1 AND 2, "CLOSE/PURGE" CONTROL

■ REFERENCE DESIGNATORS: 81V76A111K19

81V76A111K20

82V76A112X21 82V76A112K22

82V76A112K25 82V76A112K26

83V76A113K21

B3V76A113K22

■ QUARTITY OF LIKE ITEMS: 8 EIGHT (FOUR PER SIDE)

FUNCTION:

UPON RECEIVING A STIMULUS FROM ASSOCIATED HYBRID DRIVERS OR BY DIRECT COMMAND FROM A FLIGHT MOM. CONTACT SETS OF TWO HYBRID RELAYS IN SERIES COMMECT 3 PHASE AC VOLTAGE TO MOTORS TO OPERATE THE LEFT AND RIGHT FORWARD VENT DOORS 1 AND 2 TO THE CLOSED OR PURGE POSITION.

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FAILURE MODES EFFECTS ANALYSIS (FMEA) CRITICAL FAILURE MODE NUMBER: 05-6AB-2126-03								
	ŔEVISION#	Z	07/20/90	R				
	SUBSYSTEM: EPOGC - ACTUATOR, VENT DOORS LRU :FWD MCA-1 ITEM NAME: RELAY, HYBRID	CRI F	TICALITY O	F THIS				
•	FAILURE MODE: SHORTS CONTACT-TO-CONTACT (PHASE "B" OR PHASE "C")							
	MISSION PHASE: DO DE-ORBIT							
•	VEHICLE/PAYLOAD/KIT EFFECTIVITY: 102 COLUMBIA : 103 DISCOVERY : 104 ATLANTIS							
•	CAUSE: PIECE PART FAILURE, CONTAMINATION, VIBRATION, MECHAN PROCESSING ANOMALY, THERMAL STRESS	NICAL S	носк,					
=	CRITICALITY 1/1 DURING INTACT ABORT ONLY? HO							
•								
•	PASS/FAIL RATIONALE: A)							
	■ B) FAILS SCREEN "B" BECAUSE RELAY SHORTS CONTACT-TO-CONTACT IS NOT READILY DETECTABLE INFLIGHT.							
•	C)							
	- FAILURE EFFECTS -							
•	(A) SUBSYSTEM: FIRST FAILURE - NO EFFECT							
•	(B) INTERFACING SUBSYSTEM(S): FIRST FAILURE - NO EFFECT							

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FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL FAILURE MODE NUMBER: 05-6AB-2126-03

- (C) MISSION: FIRST FAILURE - NO EFFECT
- (0) CREW, VEHICLE, AND ELEMENT(S): FIRST FAILURE - NO EFFECT
- (E) FUNCTIONAL CRITICALITY EFFECTS:
 - 1. HYBRID RELAY SHORTS CONTACT TO CONTACT (EITHER PHASE "8" OR PHASE "C")
 - 2. SERIAL HYBRID RELAY SHORTS CONTACT-TO-CONTACT ON SIMILAR PHASE
 - LOSS OF REDUNDANT MOTOR
 - 4. REDUNDANT DOOR FAILS CLOSED

AFTER THE SECOND FAILURE, PHASE-TO-PHASE SHORT WOULD OCCUR WHEN OPEN COMMAND IS PRESENT CAUSING AC CIRCUIT BREAKER TO TRIP WHICH RESULTS IN LOSS OF ABILITY TO OPEN VENT DOOR VIA ASSOCIATED MOTOR. POSSIBLE LOSS OF CREW/VEHICLE DUE TO INABILITY TO PROVIDE SUFFICIENT VENTING IF BOTH VENT DOORS FAIL CLOSED RESULTING IN STRUCTURAL OVERLOAD DURING ENTRY.

- DISPOSITION RATIONALE -

- (A) DESIGN: REFER TO APPENDIX C, ITEM NO. I - HYBRID RELAY
- (B) TEST: REFER TO APPENDIX C. ITEM NO. 1 - HYBRID RELAY

GROUND TURNAROUND TEST NO OMRSD TEST AVAILABLE

- (D) FAILURE HISTORY: REFER TO APPENDIX C. ITEM NO. 1 - HYBRID RELAY
- (E) OPERATIONAL USE: NONE

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FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL FAILURE MODE

NUMBER: 05-648-2126-03

- APPROVALS -

RELIABILITY ENGINEERING: T. AI
DESIGN ENGINEERING : J. KRAGER
QUALITY SUPERVISOR : J. COURSEN

NASA RELIABILITY

NASA SUBSYSTEM MANAGER :

NASA SUBSYSTEM MANAGER :

NASA EPO&C RELIABILITY : NASA QUALITY ASSURANCE :

8/24/90