FAILURE MODES EFFECTS ANALYSIS (FMEA) - CIL HARDWARE

NUMBER: 05-68A-2207 -X

SUBSYSTEM NAME: EPD&C - LANDING GEAR CONTROL

REVISION: 2

02/13/90

PART DATA

PART NAME

VENDOR NAME

PART NUMBER

VENDOR NUMBER

LRU

: FWD PCA 2

V070-763340

LRU

: FWD PCA 3

V070-763360

SRU

: DIODE

JANTX1N1204RA

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:

DIODE, BLOCKING (12A), POWER, REDUNDANT SHUTOFF VALVE

REFERENCE DESIGNATORS:

82V76A23CR15

83V76A24CR10

QUANTITY OF LIKE ITEMS: 2

TWO, ONE IN FPCA-2 AND ONE IN FPCA-3

FUNCTION:

ISOLATES TWO DIFFERENT MAIN DC BUS VOLTAGES AT THE RPC'S OUTPUT FOR POWERING THE REDUNDANT SHUTOFF VALVE IN PARALLEL. ONE DIODE AT OUTPUT OF EACH RPC FOR REDUNDANT MAIN-DC POWER TO REDUNDANT SHUTOFF VALVE.

FAILURE MODES EFFECTS ANALYSIS FMEA — CIL FAILURE MODE

NUMBER: 05-88A-2207-01

REVISION#:

3

06/28/99

SUBSYSTEM NAME: EPD&C - LANDING GEAR CONTROL

LRU: FWD PCA 2

CRITICALITY OF THIS

ITEM NAME: DIODE

FAILURE MODE: 1R3

FAILURE MODE:

OPEN, FAILS TO CONDUCT

MISSION PHASE:

DO DE-ORBIT

VEHICLE/PAYLOAD/KIT EFFECTIVITY:

102 COLUMBIA

103 DISCOVERY

104 ATLANTIS

105 ENDEAVOUR

CAUSE:

STRUCTURAL FAILURE (MECHANICAL STRESS, VIBRATION), ELECTRICAL STRESS,

THERMAL STRESS, PROCESSING ANOMALY

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN

A) PASS

B) FAIL

C) PASS

PASS/FAIL RATIONALE:

A)

B)

FAILS "B" SCREEN BECAUSE DIODE FAILURE IS NOT FLIGHT DETECTABLE.

C)

- FAILURE EFFECTS -

(A) SUBSYSTEM:

FIRST FAILURE - LOSS OF CAPABILITY TO CONDUCT POWER FROM ONE OF TWO PATHS

(B) INTERFACING SUBSYSTEM(S);

FIRST FAILURE - LOSS OF REDUNDANT POWER TO REDUNDANT SHUTOFF VALVE

FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CIL FAILURE MODE NUMBER: 05-6BA-2207- 01

(C) MISSION:

FIRST FAILURE - NO EFFECT

(D) CREW, VEHICLE, AND ELEMENT(S):

FIRST FAILURE - NO EFFECT

(E) FUNCTIONAL CRITICALITY EFFECTS:

POSSIBLE LOSS OF CREWIVEHICLE DUE TO LOSS OF ELECTRICAL POWER TO REDUNDANT SHUT-OFF VALVE PLUS LOSS OF ALL ASSOCIATED REDUNDANCY (LOSS OF LANDING GEAR DUMP VALVE AND RETRACT/CIRC VALVE) FOR EXTENDING THE LANDING GEARS.

-DISPOSITION RATIONALE-

(A) DESIGN:

REFER TO APPENDIX F, ITEM NO. 2 - DIODE

(B) TE\$T;

REFER TO APPENDIX F, ITEM NO. 2 - DIODE

GROUND TURNAROUND TEST

ANY TURNAROUND CHECKOUT TESTING IS ACCOMPLISHED IN ACCORDANCE WITH OMRSD.

(C) INSPECTION:

REFER TO APPENDIX F. ITEM NO. 2 - DIODE

(D) FAILURE HISTORY:

CURRENT DATA ON TEST FAILURES, FLIGHT FAILURES, UNEXPLAINED ANOMALIES, AND OTHER FAILURES EXPERIENCED DURING GROUND PROCESSING ACTIVITY CAN BE FOUND IN THE PRACA DATA BASE.

(E) OPERATIONAL USE:

NONE

FAILURE MODES EFFECTS ANALYSIS (FMEA) - CIL FAILURE MODE NUMBER: 05-6BA-2207-01

- APPROVALS -

EDITORIALLY APPROVED TECHNICAL APPROVAL

: BNA

: VIA APPROVAL FORM

: J. Xumura 7/6/49 : 98-CIL-011_05-6BA(2)