

SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : EPD&C - MAIN PROP. FMEA NO 05-6J -2130 -1 REV:06/15/88

ASSEMBLY : PANEL R2 CRIT. FUNC: 1R
 P/N RI : ME451-0018-0100 CRIT. HDW: 2
 P/N VENDOR: VEHICLE 102 103 104
 QUANTITY : 1 EFFECTIVITY: X X X
 : ONE PHASE(S): PL LO X OC DO LS
 :

REDUNDANCY SCREEN: A-PASS B-PASS C-PASS
 PREPARED BY: APPROVED BY: APPROVED BY (NASA):
 DES J BROWN DES [Signature] EPDC SSM [Signature]
 REL F DEFENSOR REL [Signature] 6/27/88 MPS SSM [Signature]
 QE D M MASAI QE [Signature] 6/27/88 EPDC REK [Signature]
 MPS REV [Signature]

ITEM:
 FUSE (1 AMP), LEFT ENGINE PNEUMATIC HELIUM CROSSOVER VALVE POWER CONTROL.

FUNCTION:
 PROVIDES CONTROL BUS PROTECTION IN THE EVENT OF LEFT ENGINE PNEUMATIC HELIUM CROSSOVER VALVE CONTROL CIRCUIT FAULTS. 32V73A2F76.

FAILURE MODE:
 FAILS OPEN, FAILS TO CONDUCT.

CAUSE(S):
 STRUCTURAL FAILURE, CONTAMINATION, VIBRATION, MECHANICAL SHOCK, PROCESSING ANOMALY, THERMAL STRESS.

EFFECT(S) ON:
 (A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE (E) FUNCTIONAL CRITICALITY
 (A) LOSS OF MANUAL SWITCH COMMAND.
 (B) LOSS OF MANUAL CAPABILITY TO CONTROL LEFT ENGINE PNEUMATIC CROSSOVER VALVE.
 (C,D) NO EFFECT - FIRST FAILURE.

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- (E) 1R/2, 1 SUCCESS PATH AFTER FIRST FAILURE.
TIME FRAME - ENGINE OPERATION.
1) HELIUM LEAK DOWNSTREAM OF CHECK VALVE CV9 (ASSUMES LEAK RATE IS LARGE ENOUGH TO DEplete PNEUMATIC SUPPLY BUT LESS THAN REQUIRED TO OVERPRESSURIZE THE AFT COMPARTMENT).
2) FUSE FAILS OPEN, CAUSING HELIUM CROSSOVER VALVE TO REMAIN CLOSED WHEN CREW MANUALLY COMMANDS LV10 OPEN IN RESPONSE TO FIRST FAILURE PRIOR TO MECO.

LOSS OF PNEUMATIC ACTUATION HELIUM RESULTS IN LO2 PREVALVE FAILING TO CLOSE AND INABILITY TO MAINTAIN INJECTED HELIUM AND LO2 PRESSURE TO THE HIGH PRESSURE OXYGEN TURBOPUMP TO PREVENT PUMP OVERSPEED AND CAVITATION AT MECO. RESULTS IN UNCONTAINED ENGINE DAMAGE, AFT COMPARTMENT OVERPRESSURIZATION, AND FIRE/EXPLOSION HAZARD. POSSIBLE LOSS OF CREW/VEHICLE.

DISPOSITION & RATIONALE:

(A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE

(A-D) FOR DISPOSITION AND RATIONALE:

REFER TO APPENDIX D, ITEM NO. 4 - FUSE, PLUG-IN TYPE.

(B) GROUND TURNAROUND TEST

COMPLETE ELECTRICAL VERIFICATION, V41AAG.160 EVERY FLIGHT.

(E) OPERATIONAL USE

NO CREW ACTION CAN BE TAKEN.

05-6J-235