## SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : EPD&C - AFT-RCS FMEA NO 05-6KA-2011 -1 REV:11/03/87

ASSEMBLY : PANEL A14

:ME451-0009-1001

P/N VENDOR: QUANTITY

P/N RI

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2 CRIT. FUNC: HDW: 7

CRIT. 104

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EFFECTIVITY: PHASE(S): PL X LO X OO X DO

REDUNDANCY SCREEN:

A-APPROVED

B-

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PREPARED BY:

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J BEEKMAN

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APPROVED BY:

Maria CL How 11-19-87

VEHICLE

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ITEM:

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FUSE (1 AMP) - LEFT AND RIGHT AFT RCS PODS THRUSTER HEATER CONTROL, MANIFOLDS 1 THROUGH 4.

## FUNCTION:

CONDUCTS (MANIFOLD 1 THROUGH 4) HEATER CONTROL CIRCUIT CURRENT AND PROVIDES PROTECTION OF CIRCUITRY. 36V73A14F27,28,29,30.

## FAILURE MODE:

OPEN, INADVERTENTLY OPENS.

## CAUSE(S):

CONTAMINATION, CHEMICALLY DEGRADED MATERIAL, STRUCTURAL FAILURE.

# EFFECT(S) ON:

- (A) SUBSISTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE
- (A) LOSS OF VOLTAGE TO THE ASSOCIATED PANEL SWITCHES.
- (B) LOSS OF ABILITY TO ENABLE THE RELATED THRUSTER HEATERS. POSSIBLE LOSS OF REDUNDANCY IF LOW TEMPERATURE AFFECTS THRUSTER OPERATION. NO EFFECT, OTHER REDUNDANT THRUSTERS ARE AVAILABLE TO COMPLETE THE REQUIRED FUNCTIONS.
- (C) PERIODIC HOT FIRING OF PRIMARY THRUSTERS WILL BE REQUIRED TO MAINTAIN THE THRUSTER TEMPERATURES ABOVE THE MINIMUM REQUIREMENT. CORRECTIVE ACTION COULD DISRUPT MISSION OBJECTIVES. MICROGRAVITY EXPERIMENTS ARE AFFECTED BY HOT FIRES. INCREASE PROPELLANT USAGE IS REQUIRED TO MAINTAIN TEMPERATURES.
- (D) NO EFFECT.

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#### DISPOSITION & RATIONALE:

(A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE

(A-D) FOR DISPOSITION AND RATIONALE REFER TO APPENDIX D, ITEM NO. 2 - FUSE, AXIAL LEAD CARTRIDGE.

#### (B) GROUND TURNAROUND TEST

COMPONENT CHECKED OUT EVERY FLIGHT DURING GROUND TURNAROUND. THE TESTING CONSISTS OF CYCLING VALVE MANUAL SWITCHES AND/OR SENDING GENERAL PURPOSE COMPUTER (GPC) COMMANDS TO CYCLE VALVES OR HEATERS WHILE MONITORING VEHICLE INSTRUMENTATION TO DETERMINE IF COMPONENTS HAVE FAILED.

## (E) OPERATIONAL USE

HOT FIRE JET AS REQUIRED TO MAINTAIN THRUSTER TEMPERATURE IN ACCEPTABLE RANGE. SOME MISSION OBJECTIVES MAY NOT BE MET DUE TO HIGHER PROPELLANT CONSUMPTION CAUSED BY PERIODIC PRIMARY THRUSTER HOT FIRE. MICROGRAVITY EXPERIMENTS WILL BE DISRUPTED DUE TO HIGHER ACCELERATION RATE OF PRIMARY THRUSTERS.