SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : EPD&C - FWD-RCS

FMEA NO 05-6KF-2038 -1 REV:11/03/87

2

ASSEMBLY : PANEL A14

P/N RI :ME452-0102-7206 CRIT. FUNC: CRIT. HDW:

P/N VENDOR: QUANTITY :1

VEHICLE

102 103 104

ONE

EFFECTIVITY: PHASE(S): PL

X TO X OO X DO

PREPARED BY:

REDUNDANCY SCREEN: A-APPROVED BY:

OΕ

APPROVED BY (NASA) >

B-C-

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RELAK LEGIT SHERRESTE

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ITEM:

TOGGLE SWITCH (2P3T) HERMETIC SEAL - PRESSURIZATION/PROPELLANT SYSTEM-HEATER SELECT.

FUNCTION:

PROVIDES THE CREW WITH THE CAPABILITY TO SELECT BETWEEN REDUNDANT SYSTEMS "AUTO A/AUTO 3", OR "OFF" FOR OPERATING THE PRESSURIZATION/PROPELIANT SYSTEM HEATERS. CIRCUITS MUST BE COMPLETED THROUGH BOTH SWITCH LEGS, IN EITHER AUTO POSITION, BEFORE HEATER CIRCUITS CAN BE ENERGIZED. 36V73A14S3.

FAILURE MODE:

FAILS TO CONDUCT, FAILS TO CLOSE, INADVERTENTLY OPENS (ONE OR MORE CONTACT SETS).

CAUSE(S):

PIZCE PART STRUCTURAL FAILURE, CONTAMINATION.

EFFECT(S) ON:

- (A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE
- (A) HEATER CIRCUIT FOWER RELAY OR THERMAL SWITCH LOGIC CIRCUIT IS NOT ENERGIZED.
- (B) LOSS OF POWER OR TURN-ON CONTROL OF PRESSURIZATION/PROPELLANT REATER SYSTEMS.
- (C) POSSIBLE MISSION MODIFICATION OR EARLY MISSION TERMINATION. FAILURE COULD DISABLE ALL OF THE FORWARD RCS MANEUVERING CAPABILITY DUE TO LOW SYSTEM TEMPERATURE.
- (D) NO EFFECT.

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DISPOSITION & RATIONALE:

- (A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE
- (A-D) FOR DISPOSITION AND RATIONALE REFER TO APPENDIX A, ITEM NO. 1 -TOGGLE SWITCH.
- (B) GROUND TURNAROUND TEST COMPONENT CHECKED OUT EVERY FLIGHT DURING GROUND TURNAROUND. THE TESTING CONSISTS OF CYCLING VALVE MANUAL SWITCHES AND/OR SENDING GENERAL PURPOSE COMPUTER (GPC) COMMANDS TO CYCLE VALVES OR HEATERS WHILE MONITORING VEHICLE INSTRUMENTATION TO DETERMINE IF COMPONENTS HAVE FAILED.
- (E) OPERATIONAL USE IF FORWARD RCS PROPELLANT LINES BECOME TOO COLD FOR USE, REDLINE ADDITIONAL AFT RCS PROPELLANT FOR TAIL-ONLY ATTITUDE CONTROL. MAY RESULT IN EARLY MISSION TERMINATION.