## CRITICAL ITEMS LIST (CIL)

SYSTEM: ASI FUNCTIONAL CRIT: 1
SUBSYSTEM: Electrical Cable Trays PHASE(S): b
REV & DATE: K, 6-29-01 HAZARD REF: S.11

DCN & DATE: 001, 2-7-02

ANALYSTS: J. Hicks/E. Howell

FAILURE MODE: Structural Failure

FAILURE EFFECT: b) Loss of mission and vehicle/crew due to LO2 tank structural failure

or debris source to Orbiter from cable tray cover.

TIME TO EFFECT: Immediate

 ${\tt FAILURE\ CAUSE(S):} \qquad {\tt A:} \qquad {\tt Improper\ Manufacture}$ 

B: Failure of Attaching Hardware

REDUNDANCY SCREENS: Not Applicable

FUNCTIONAL DESCRIPTION: Provide environmental protection for lines and cables routed along

the LO2 tank surface.

FMEA ITEM	PART NO.	PART NAME	QTY	EFFECTIVITY
4.3.1.1	80911041206-039	Cable Tray Cover Assy (LO2)	1	ET-61 & Up
4.3.2.1	80911041206-009	Cable Tray Cover Assy (LO2)	5	ET-61 & Up
4.3.3.1	80911041206-010	Cable Tray Cover Assy (LO2)	4	ET-61 & Up
	80911041206-010 80911017063-009	Cable Tray Cover Assy (LO2) Cable Tray Cover Assy (LO2)	3 1	ET-115 ET-115

REMARKS: The cover assemblies are grouped as the failure mode, causes and effects are the same.

MASTER 4.3-1

## CRITICAL ITEMS LIST (CIL)

CONTINUATION SHEET

SYSTEM: ASI REV & DATE: K, 6-29-01 SUBSYSTEM: Electrical Cable Trays DCN & DATE: 001, 2-7-02

FMEA ITEM CODE(S): 4.3.1.1, 4.3.2.1, 4.3.3.1

## RATIONALE FOR RETENTION

#### DESIGN:

- A, B: The covers are machined from 2024-T81 aluminum alloy sheet stock. Materials selected for this part number are in accordance with MMC-ET-SE16 which assures repetitive conformance of composition and properties.
- A: The covers are designed to the required yield (1.1) and ultimate (1.4) safety factors (ET Stress Report 826-2188).
- B: The attaching hardware is selected from the Approved Standard Parts List (ASPL 826-3500). The hardware is installed per STP2014 and torqued using values specified on Engineering drawings. Tensile installation loads are sufficient to provide screening for major flaws in individual fasteners.

#### TEST:

The Cable Tray Cover Assembly (LO2) is certified. Reference HCS MMC-ET-TM08-L-S015 (LWT-54 thru 88) and HCS MMC-ET-TM08-L-S510 (LWT-89 & Up).

#### Vendor:

B: Attaching fasteners are procured and tested to standard drawings NAS1219, 26L17 and 34L1.

#### INSPECTION:

#### Vendor Inspection-Lockheed Martin Surveillance:

- A, B: Verify materials selection and verification controls (MMC-ET-SE16, drawing 80911041206 and standard drawings 26L17, NAS1219 and 34L1).
- A: Inspect dimensional conformance (drawing 80911041206).

## MAF Quality Inspection:

- B: Inspect that attaching hardware is free from damage (drawing 80911041205 and STP2014).
- A, B: Verify installation and witness torque (drawing 80911041205).
- B: Verify locking feature (drawing 80911041205 and STP2014).
- A: Inspect dimensional conformance (drawing 80911017063) (Effectivity ET-115).

# Launch Site Inspection:

- B: Inspect that attaching hardware is free from damage (drawing 80931017071 and STP2014)(Effectivity ET-115)
- A, B: Verify installation and witness torque (draing 80931017071) (Effectivity ET-115).
- B: Verify locking feature (drawing 80931017071 and STP2014)( Effectivity ET-115).

#### FAILURE HISTORY:

Current data on test failures, unexplained anomalies and other failures experienced during ground processing activity can be found in the PRACA data base.

MASTER 4.3-2