

July 31, 1994

CRITICAL ITEMS LIST

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1) CIC ITEM : A15D-03  
2) FMEA CODE : A15D  
3) COMPONENT : HEAT EXCHANGER  
4) PART NUMBER : RS00601  
5) SYSTEM/SUBSYSTEM : COMBUSTION DEVICES/AXXX  
6) FAILURE MODE : BYPASS LINE ORIFICE RESTRICTION

7) PREPARED : SSNE RELIABILITY  
8) APPROVED : *[Signature]*  
9) DATE : 07-31-94  
10) REVISION/CHANGE : -003/0  
11) EFFECTIVITY : -451  
12) HAZARD REFERENCE : SEE LISTINGS BELOW  
13) COBD #: ME3-01-1679

PHASE	FAILURE DESCRIPTION/EFFECT	CRITICALITY
SNC	HIGH TEMPERATURE GOX WOULD BE DELIVERED TO THE POGO ACCUMULATOR AND ORBITER TANK PRESSURIZATION SYSTEM CAUSING ENGINE AND VEHICLE HARDWARE FAILURE. LOSS OF VEHICLE.  REDUNDANCY SCREENS: SINGLE POINT FAILURE: N/A	1 HAZARD REF: ME-055

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RSS-8740-11

CIL ITEM: A150-03		DESIGN	DOCUMENT REF.
FAILURE CAUSE A: CONTAMINATION, FOREIGN MATERIAL			
<p>A MULTIPLE INLET SINGLE OUTLET (MISO) HEAT EXCHANGER BYPASS ORIFICE CONTROLS THE AMOUNT OF LOX THAT COMBINES WITH THE LOX LEAVING THE HEAT EXCHANGER (1). THE MISO BYPASS ORIFICE IS DESIGNED FOR MINIMUM SUSCEPTABILITY TO FLOW BLOCKAGE BY PROVIDING REDUNDANT FLUID ENTRANCE PATHS. THIS FEATURE ALLOWS THE ORIFICE TO TOLERATE GROSS CONTAMINATION WITHOUT SIGNIFICANT FLOW RESTRICTION. THE ORIFICE IS DESIGNED TO ALLOW PARTICLES LESS THAN OR EQUAL TO 1500 MICRONS TO PASS THROUGH. THE LOX SUPPLY SYSTEM IS DESIGNED TO FILTER LOX IN TWO STEPS, AN 800 MICRON FILTER AT THE E.T. INTERFACE (2) AND A 100 MICRON FILTER AT THE RTI FLOOD VALVE (3). THE FILTER IS CONICAL SHAPED TO GIVE MORE SURFACE AREA TO PREVENT BLOCKAGE (4). THE FILTER DESIGN ALLOWS FOR TRAPPED PARTICLES TO SETTLE OFF OF THE FILTERING AREA. THE FILTERS ARE CERTIFIED TO FILTERING ABILITY (5).</p>			(1) R035519 (2) 100 13415000 (3) R8007083 (4) 286-5008 (5) RC286-5009
CIL ITEM: A150-03		INSPECTION AND TEST	
POSSIBLE CAUSES	SIGNIFICANT CHARACTERISTICS	INSPECTION(S)/TEST(S)	DOCUMENT REF.
FAILURE CAUSE A:	R035519 - BYPASS ORIFICE 286-5008-001 - FILTER		R035519 286-5008
	LOX FILTERING	UPSTREAM COMPONENTS ARE CLEANED TO DRYGEN SERVICE OR BETTER REQUIREMENTS.  BYPASS LINES AND ASSEMBLIES ARE CLEANED AND INSPECTED TO LOX SERVICE SPECIFICATIONS.  ANTI-FLOOD VALVE FILTER REPLACEMENT IS VERIFIED PRIOR TO EVERY FLIGHT. (LAST TEST)	RL10001  RL10001  UMPS0 V41BL0.220
	ASSEMBLY INTEGRITY	THE HOT FIRE TESTING AND 2ND E & M VERIFIES PASSAGES ARE CLEAR.	RL00050-04 RL00056-06 RL00056-07
	DATA REVIEW	ALL CONTROLLER DATA FROM THE PREVIOUS FLIGHT OR HOT-FIRE IS REVIEWED. ANY ANOMALOUS CONDITION NOTED REQUIRES FURTHER TESTING OR HARDWARE REPLACEMENT PRIOR TO NEXT FLIGHT.	MSFC PLR 1220
FAILURE HISTORY: NONE			

OPERATIONAL USE: NOT APPLICABLE

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RSS-8740-11

SSWIC    WAGIL  
**WELD JOINTS**

Component Group: Combustion Devices  
 CIL Item: A150  
 Component: RS006801  
 Part Number: Heat Exchanger  
 A150

Prepared: A. Kay  
 Approved: T. Nguyen  
 Approval Date: 9/9/99  
 Change #: 1  
 Directive #: CCBD ME3-01 5238  
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Component	Basic Part Number	Weld Number	Weld Type	Class	Root Side Not Access	Critical Initial Flaw Size Not Detectable		Comments
						HCF	LCF	
HEX BYPASS LINE	RS006895	1,2	GTAW	I		X		
HEX OUTLET MANIFOLD	RS006895	1,2	GTAW	I		X		
HEX OUTLET MANIFOLD	RS006895	3,4	GTAW	I				
HEX OXIDIZER TANK	RS006801	1	GTAW	II	X			
HEX OXIDIZER TANK	RS006801	2	GTAW	II				
HEX OXIDIZER TANK	RS006801	2	GTAW	I	X	X	X	(A350)
HEX OXIDIZER TANK	RS006801	4	GTAW	I	X	X		
HEX OXIDIZER TANK	RS006801	5	GTAW	I	X	X		
HEX OXIDIZER TANK	RS006801	6,7	GTAW	I	X			
HEX OXIDIZER TANK	RS006801	9(OPT)	GTAW	I				
HEX OXIDIZER TANK	RS006801	8(OPT)	ESW	I				
HEX OUTER SHELL	RS006802	6(OPT)	ESW	I				(A350)
HEX OUTER SHELL	RS006802	6(OPT)	GTAW	I				(A350)
HEX OXIDIZER TANK LINER	RS006806	1-9	GTAW	II	X			
HEX OXIDIZER TANK LINER	RS006806	20-23,25- 66,72-87	GTAW	II	X			
HEX OXIDIZER TANK LINER	RS006806	24,70	GTAW	I	X			
HEX OXIDIZER TANK LINER	RS006806	68,71	GTAW	I				
HEX OXIDIZER TANK LINER	RS006806	88-95	GTAW	II	X			
HEX OXIDIZER TANK LINER	RS006806	98-106	GTAW	II	X	X		
BRACKET	RS006810	1,2	GTAW	II	X			
BRACKET	RS006810	3,4	ERW	II				
BRACKET	RS006810	9,10	GTAW	II				
HEX INNER SHELL	RS006811	1	GTAW	I				
HEX INNER SHELL	RS006811	2,3	GTAW	I				
HEX COIL	RS006812	1,2	GTAW	I				
HEX COIL	RS006812	3	GTAW	I				
HEX COIL	RS006812	4	GTAW	I				

SSME - A/CIL  
**FIELD CONFIGURATION VARIANCES FROM CIL RATIONALE**

Component Group: Combustion Devices  
 Item Name: Heat Exchanger  
 Item Number: A150  
 Part Number: RS008801

Prepared: A. Key  
 Approved: T. Nguyen  
 Approval Date: 9/9/99  
 Change #: 1  
 Directive #: CCB ME3-01-5238

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Base Line Rationale	Variance	Change Rationale	Variant Dash Number
1. A150-01 TUBE WELDS MEASURED TO VERIFY WFLD WIDTH AND CENTERED ON JOINT (QE-86-229)	NO WELD POSITION MEASUREMENT.	MEASUREMENT ASSURED THAT WELDER COMPLETELY FUSED JOINT, WHICH PROVIDES HIGH CONFIDENCE THAT NO SIGNIFICANT WELD DEFECTS EXIST. USE AS IS RATIONALE: 1. WELD WIDTHS READ FROM ORIGINAL X-RAYS ADD CONFIDENCE THAT JOINT WAS COMPLETELY FUSED. 2. THIN WALL ALLOWS HIGH RESOLUTION X-RAYS, AND X-RAY ENHANCEMENT AND INTERPRETATION BY LEVEL 3 NDE GIVES ADDED ASSURANCE OF NO SIGNIFICANT DEFECTS. 3. ETCH AND TYPE IVC PENETRANT AFTER PROOF AND HOT FIRE OF RS008811 WELD 1 ASSURES NO SIGNIFICANT I.D. DEFECTS. 4. HIGH RESOLUTION I.D. BORESCOPE INSPECTION AFTER PROOF AND HOT FIRE OF RS008811 WELD 1 AND RS008812 WELD 3 PROVIDES ADDITIONAL CONFIDENCE THAT NO SIGNIFICANT I.D. DEFECTS EXISTS. 5. X-RAY AFTER PROOF AND HOT FIRE OF RS008812 WELDS 1, 2, AND 3 ADDS CONFIDENCE THAT NO SIGNIFICANT DEFECTS EXIST.	401 -441
2. A150-01 TUBE WELDS ARE PENETRANT INSPECTED BEFORE REAMING (QE-86-229)	NO PENETRANT BEFORE REAM OF PRIMARY TUBE WELD I.D.s.	PROVIDED ADDITIONAL CONFIDENCE THAT I.D. SURFACE DEFECTS DO NOT EXIST. USE AS IS RATIONALE: 1. THIN WALL ALLOWS HIGH RESOLUTION X-RAYS, AND X-RAY ENHANCEMENT AND INTERPRETATION BY LEVEL 3 NDE GIVES ADDED ASSURANCE OF NO SIGNIFICANT DEFECTS 2. ETCH AND TYPE IVC PENETRANT AFTER PROOF AND HOT FIRE OF RS008811 WELD 1 ASSURES NO SIGNIFICANT I.D. EFFECTS 3. HIGH RESOLUTION I.D. BORESCOPE INSPECTION AFTER PROOF AND HOT FIRE OF RS008811 WELD 1 AND RS008812 WELD 3 PROVIDES ADDITIONAL CONFIDENCE THAT NO SIGNIFICANT I.D. DEFECTS EXIST. 4. X-RAY AFTER PROOF AND HOT FIRE OF RS008812 WELD 3 ADDS CONFIDENCE THAT NO SIGNIFICANT DEFECTS EXIST. 5. LEAK TEST PRIOR TO EACH FLIGHT WILL DETECT THROUGH CRACKS. 6. RS008811 WELD 3 I.D. IS PENETRANT INSPECTED AFTER REAMING, BUT WITHOUT ETCHING	-401 -441

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