

Component Group:

Propellant Valves

CfL Item: Component: Part Number:

D110-04 Main Fuel Velve RS008258

Fallure Mode:

Structural failure.

Prepared:

Approved: Approval Date: Change #:

T. Nguyen 6/30/99

Directive #:

CCBD ME3-01-5226

P. Lowrimore

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Phase	Failure / Effect Description		Criticality Hazard Reference
PSMCD	Fuel flow to combustors reduced; high pressure fuel leakage into aft compartment. Loss of vehicle.		1
4.1	Redundancy Screens: SINGLE POINT FAILURE: N/A.		ME-D3P,D, ME-D3S,A,M,C

## SSME / TA/CIL <u>DE</u>کینام

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Design / Document Reference

## FAILURE CAUSE: A: Fracture of housing or end cap.

THE MFV HOUSING IS MACHINED FROM A 5AL-2.5SN (ELI) TITANIUM FORGING (1). THE TITANIUM ALLOY IS CORROSION RESISTANT, AND IS RESISTANT TO STRESS CORROSION CRACKING AT THE OPERATING TEMPERATURE AND EXPOSURE CONDITIONS. 5AL-2.5SN (ELI) TITANIUM IS RESISTANT TO HYDROGEN ENVIRONMENT EMBRITH. EMENT AT THE NEAR -400 DEGREES F OPERATING ENVIRONMENT. THE VALVES SHORT EXPOSURE TO GASEOUS HYDROGEN MAKES HYDRIDE FORMATION NOT A PROBLEM (2). THE ROUGH MACHINED HOUSING IS STRESS RELIEVED AFTER PRESSURE LOADING (3) HIGH CYCLE AND LOW CYCLE FATIGUE LIFE OF THE MFV CAP AND HOUSING MEET CEI REQUIREMENTS (4). MINIMUM FACTORS OF SAFETY FOR THE CAP AND HOUSING MEET CEI REQUIREMENTS (5). A SPECIALLY STRAIN GAGED HOUSING WAS HYDROSTATICALLY PRESSURIZED AND STRAINS WERE RECORDED. THE HOUSING DEMONSTRATED LOW STRAINS AT ALL STRAIN GAGE LOCATIONS (6). THE MFV HOUSING AND CAP WERE CLEARED FOR FRACTURE MECHANICS/NDE FLAW GROWTH BY CRITICAL INITIAL FLAW SIZE DETECTABILITY (7). THE VALVE COMPLETED DVS TEST REQUIREMENTS (8) INCLUDING ENDURANCE (9) AND VIBRATION (10).

(t) RS008207; (2) RSS-8576; (3) RS008201, RS008275; (4) RL00532, CP320R0003B; (5) RSS-8546, CP320R0003B; (6) SSME-82-142; (7) NASA TASK 117; (8) DVS-SSME-515; (9) RSS-515-17; (10) RSS-515-24

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	Failure Causes	Significant Characteristics	Inspection(s) / Test(s)	Document Reference
A		HOUSING CAP ASSEMBLY		RS008201 R\$008275
		MATERIAL INTEGRITY	THE MATERIAL IS VERIFIED PER DRAWING REQUIREMENTS.	
			THREE STANDARD TEST SPECIMENS ARE TAKEN FROM THE THREE AXES OF EACH FORGING AND AT LEAST ONE SPECIMEN FROM EACH AXIS IS TESTED TO ESTABLISH CONFORMANCE WITH THE REQUIRED MECHANICAL PROPERTIES. TRACEABILITY DATA ARE RECORDED FOR EACH FORGING.	RS008207
			THE HOUSING IS PRESSURE TESTED AFTER ROUGH MACHINING.	RL00440
			THE HOUSING AND CAP ARE PROOF PRESSURE TESTED AT CRYOGENIC TEMPERATURES AFTER FINAL MACHINING.	RL00438 RL00439
		THE HOUSING AND CAP ARE PERETRANT INSPECTED AFTER FINAL MACHINING.	THE HOUSING AND CAP ARE PENETRANT INSPECTED AFTER FINAL MACHINING.	RA0115-116
		HEAT TREAT	STRESS RELIEF OF HOUSING AFTER ROUGH MACHINING AND PRESSURE TEST IS VERIFIED PER SPECIFICATION REQUIREMENTS.	RA0111-024
		ASSEMBLY INTEGRITY	THE ASSEMBLED VALVE IS PROOF PRESSURE TESTED AND FUNCTIONAL CHECKOUT TESTED INCLUDING LEAK CHECKS.	RL00453
			THE VALVE IS HELIUM SIGNATURE LEAK TESTED AND VALVE OPERATION IS VERIFIED DURING PRE-LAUNCH CHECKOUTS. (LAST TESTS)	OMRSD \$00000.95 OMRSD \$00FA0.2

Fallure History:

Comprehensive failure history data is maintained in the Problem Reporting database (PRAMS/PRACA)

Reference: NASA lefter SA21/88/308 and Rocketdyne letter 88RC09761.

Operational Use

Not Applicable.



Component Group:

Propellant Valves D110

CIL Item:

Main Fuel Valve

Component: Part Number:

RS008256

Prepared:

P. Lowrimore

T. Nguyen 6/30/99

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Comment	Basic Part Number	Milatel Bl., sekas	Male Time	01	Root Side Not		
Component				Class	Access	HCF LCF	 Comments
BELLOWS	R\$008208	3.4	EBW	II	x	X	
BELLOWS	RS008208	5-8	GTAW	ı			
SHAFT	R\$008271	1,2.	EBW	II	×	x	