SSME EA/CIL REDUNDANCY SCREEN

Component Group: CIL Item:

Actuators E701-02

Part Number:

RES1008-3003 Hydraulic Filter

Component: FMEA Item:

E701

Failure Mode:

Structural failure.

Prepared: Approved: Approval Date: Change #: Directive #:

P. Lowrimore T. Nguyen 6/9/00

CCBD ME3-01-5624

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Phase	Failure / Effect Description	Criticality
M	Major hydraulic fluid leak into aft compartment; loss of hydraulic pressure; loss of actuator control; controller switches to channel B (servovalve No. 2) all actuators; failure continues controller initiates budgettle location.	Hazard Reference
4.1	(servovalve No. 2) all actuators; failure continues, controller initiates hydraulic lockup all actuators. Mission abort may result when hydraulic lockup occurs during Max Q throttling.	1R ME-E1P,S,A,M,C,D
	Redundancy Screens: SINGLE POINT FAILURE: N/A	

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E.

(1)

SSME FMEA/CIL DESIGN

Component Group:

Actuators

CIL Item:

E701-02

Part Number: Component:

RES1008-3003 Hydraulic Filter

FMEA Item:

E701

Failure Mode:

Structural failure.

Prepared:

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Désign / Document Reference

FAILURE CAUSE: A: Structural failure of housing.

THE FILTER HOUSING (1) IS MADE FROM HEAT TREATED INCONEL 718. THIS MATERIAL WAS SELECTED FOR ITS STRENGTH, ELASTIC MODULUS, DUCTILITY, CORROSION AND STRESS CORROSION RESISTANCE, AND WELDABILITY (2). THE FILTER HOUSING IS WELDED TO ROCKETDYNE WELD REQUIREMENTS (1). THE FILTER IS PROOF PRESSURE TESTED TO ASSURE STRUCTURAL INTEGRITY. THE HIGH CYCLE AND LOW CYCLE FATIGUE LIFE OF THE FILTER MEET CEI REQUIREMENTS (3). THE MINIMUM FACTORS OF SAFETY FOR THE FILTER HAS COMPLETED DESIGN VERIFICATION TESTING INCLUDING VIBRATION (6). DVS TEST RESULTS ARE DOCUMENTED (7).

(1) RES1008-3003; (2) RSS-8582; (3) RL00532, CP320R0003B; (4) RSS-8546, CP320R0003B; (5) NASA TASK 117; (6) DVS-SSME-512; (7) RSS-512-52

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SSME FI /CIL

Component Group: CIL Item:

Actuators E701-02

Part Number:

RES1008-3003 Hydraulic Filter

Component: FMEA Item:

E701

Failure Mode:

Structural failure.

Prepared:

P. Lowrimore T. Nguyen

Approved: Approval Date:

6/9/00

Change #: Directive #:

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Failure Causes	Significant Characteristics	Page:	1 of 1
4	FILTER HOUSING	Inspection(s) / Test(s)	Document Reference
MATERIAL INTEGRI WELD INTEGRITY PROOF TEST	•		11-1147
	•	MATERIAL INTEGRITY AND HEAT TREAT ARE VERIFIED PER DRAWING REQUIREMENTS.	11-1147
	WELDINIEGRIT	WELDS ARE VERIFIED PER DRAWING AND SPECIFICATION REQUIREMENTS.	11-1147
	PROOF TEST		RC1008 RL10011
		PROOF TEST VERIFIES INTEGRITY OF HOUSING PER DRAWING REQUIREMENTS.	14228-631
	<u> </u>	COMPONENT EXTERNAL INSPECTION AND AFT CLOSEOUT INSPECTIONS VERIFY NO HYDRAULIC LEAKAGE FROM FILTER STRUCTURAL HOUSING. (LAST TEST)	OMRSD V41BU0.030

Comprehensive failure history data is maintained in the Problem Reporting database (PRAMS/PRACA)

Reference: NASA letter SA21/88/308 and Rocketdyne letter 88RC09761.

Operational Use:

FAILURE MODE CAN BE DETECTED IN REALTIME BY THE FLIGHT CONTROL TEAM WHO WILL EVALUATE EFFECTS UPON VEHICLE PERFORMANCE AND ABORT CAPABILITY. BASED ON THIS EVALUATION THE APPROPRIATE ABORT MODE OR SYSTEM CONFIGURATION WILL BE SELECTED. FAILURE DETECTION CUES AND ASSOCIATED SSME PERFORMANCE DATA HAVE BEEN COORDINATED BETWEEN THE ENGINEERING AND FLIGHT OPERATIONS ORGANIZATIONS WITH THE

RESPONSES DOCUMENTED IN MISSION FLIGHT RULES.

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