## SSME EA/CIL REDUNDANCY SCREEN

Component Group:

**Ducts and Lines** 

CIL Item: Part Number: K405-01 RS007120

Component:

MFVA Hydraulic Supply Manifold

FMEA Item:

K405, K406, K407

Failure Mode:

Fails to contain hydraulic fluid.

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Prepared:

Approved: Approval Date: D. Early T. Nguyen 7/25/00

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Phase	Failure / Effect Description				
M 4.1	Hydraufic fluid leakage into aft compartment. Loss of hydraulic pressure results in hydraufic lockup. Loss of mission may result when hydraulic lockup occurs during Max Q throttling.	1R ME-E1P,S,A,M,C,D			
	Redundancy Screens: LINE SYSTEM - SENSOR SYSTEM: UNLIKE REDUNDANCY				
	A: Pass - Redundant hardware items are capable of checkout during normal ground turnaround.     B: Pass - Loss of a redundant hardware items is detectable during flight.	·			
-	C: Pass - Loss of redundant hardware items could not result from a single credible event.	<u>.</u> ,			
C 4.2	When failure occurs when MFV is less than 10%, but not closed, Channel B SEII will not occur. MFV will remain at lockup position until failsafe servoswitch is de-energized post cutoff. Open air fire if on launch pad. Loss of vehicle.	1 ME-E1P.S.A.M.C.[			
2	Redundancy Screens: SINGLE POINT FAILURE: N/A				

## SSME FMEA/CIL DESIGN

Component Group:

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Design / Document Reference

FAILURE CAUSE: A: Parent material failure or weld failure.

THE MANIFOLD ASSEMBLY (1) IS MANUFACTURED UTILIZING INCONEL 625 TUBE AND BAR. INCONEL 625 WAS SELECTED FOR ITS WELDABILITY, FORMABILITY, RESISTANCE TO STRESS CORROSION CRACKING, AND CORROSION RESISTANCE (2). INCONEL 625 POSSESSES THE REQUIRED STRENGTH WITHOUT REQUIRING HEAT TREAT. FLANGE, TEE, FITTING, AND ELBOW SECTIONS INCORPORATE RADIUS JOINTS TO REDUCE STRESS CONCENTRATIONS. OFFSET LIMIT REQUIREMENTS ARE ESTABLISHED TO REDUCE STRESS CONCENTRATIONS AND IMPROVE WELD GEOMETRY. TUBING STOCK IS PLANISHED WHEN WELDED, OR DRAWN FROM SEAMLESS TO MAINTAIN SURFACE REGULARITY. INSTALLATION IS CONTROLLED FOR ANGULARITY AND OFFSET PER SPECIFICATION REQUIREMENTS (3). MINIMUM FACTORS OF SAFETY FOR THE MANIFOLD MEET CEI REQUIREMENTS (4). HIGH AND LOW CYCLE FATIGUE LIFE MEET CEI REQUIREMENTS (5). THIS MANIFOLD ASSEMBLY WAS VERIFIED TO SATISFY PRESSURE CYCLING AND ULTIMATE PRESSURE DVS BY SIMILARITY TO THE RS007119 MANIFOLD (6). THE MANIFOLD ASSEMBLY PARENT MATERIAL WAS CLEARED FOR FRACTURE MECHANICS/NDE FLAW GROWTH, SINCE THEY ARE NOT FRACTURE CRITICAL PARTS (7). TABLE K405 LISTS ALL THE FMEA/CIL WELDS AND IDENTIFIES THOSE WELDS IN WHICH THE CRITICAL INITIAL FLAW SIZE IS NOT DETECTABLE, AND THOSE WELDS IN WHICH THE ROOT SIDE IS NOT ACCESSIBLE FOR INSPECTION. THESE WELDS HAVE BEEN ASSESSED AS ACCEPTABLE FOR FLIGHT BY RISK ASSESSMENT (8).

(1) RS007120; (2) RSS-8582; (3) RA1102-006; (4) RSS-8546, CP320R0003B; (5) RL00532, CP320R0003B; (6) SSME-80-1156; (7) NASA TASK 117; (8) RSS-8756

## SSME FM **INSPECTION AND TEST**

Component Group:

**Ducts and Lines** 

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FMEA Item:

K405, K406, K407

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Failure Causes	Cignificant Characteristics	Page:	1 of 1  Document Reference	
<del></del>	Significant Characteristics	Inspection(s) / Test(s)		
A	MANIFOLD FLANGE ELBOW FITTING TEE		RS007120 RS007234 RS007237 RS007142 RS007149	
	MATERIAL INTEGRITY	MATERIAL INTEGRITY IS VERIFIED PER DRAWING REQUIREMENTS.	RS007120 RS007234 RS007237 RS007142 RS007149	
		DETAILS ARE PENETRANT INSPECTED PER SPECIFICATION REQUIREMENTS.	RA0115-116	
		THE TUBING IS INSPECTED FOR EXCESSIVE EXTERNAL SURFACE DEFECTS.	RS007120	
	WELD INTEGRITY	ALL WELDS ARE INSPECTED TO DRAWING AND SPECIFICATION REQUIREMENTS PER WELD CLASS. INSPECTIONS INCLUDE: VISUAL, DIMENSIONAL, PENETRANT, RADIOGRAPHIC, ULTRASONIC, AND FILLER MATERIAL, AS APPLICABLE.		
	ASSEMBLY INTEGRITY	THE ASSEMBLY IS PROOF PRESSURE TESTED PER DRAWING REQUIREMENTS.	RS007120	
		WELDS ARE PENETRANT INSPECTED AFTER PROOF TEST PER SPECIFICATION REQUIREMENTS.	RA0115-116	
	FLIGHT FLOW TESTING	FOLLOWING REPAIR OR REPLACEMENT, AN EXTERNAL LEAK CHECK IS PERFORMED TO REVALIDATE THE SUBSYSTEM.	OMRSD V41GEN.575	
		DURING EXTERNAL INSPECTIONS, THE HYDRAULIC SYSTEM IS VISUALLY INSPECTED FOR LEAKAGE.	OMRSD V418U0.030	
		DURING AFT CLOSEOUT INSPECTION, ANY EVIDENCE OF PREVIOUS HYDRAULIC LEAKAGE REQUIRES FURTHER DISPOSITION. (LAST TEST)	OMRSD V41BU0.070	

Failure History:

Comprehensive failure history data is maintained in the Problem Reporting database (PRAMS/PRACA)

Reference: NASA letter SA21/88/308 and Rocketdyne letter 88RC09761.

Operational Use:

FAILURE MODE CAN BE DETECTED IN REALTIME BY THE FLIGHT CONTROL TEAM WHO WILL EVALUATE EFFECTS UPON VEHICLE PERFORMANCE AND ABORT CAPABILITY. BASED ON THIS EVALUATION THE APPROPRIATE ABORT MODE OR SYSTEM CONFIGURATION WILL BE SELECTED. FAILURE DETECTION CUES AND ASSOCIATED SSME PERFORMANCE DATA HAVE BEEN COORDINATED BETWEEN THE ENGINEERING AND FLIGHT OPERATIONS ORGANIZATIONS WITH THE

RESPONSES DOCUMENTED IN MISSION FLIGHT RULES.

## **SSME FMEA/CIL WELD JOINTS**

Component Group:

**Ducts and Lines** 

CIL Item:

K405

Part Number: Component:

RS007120

FMEA Item:

MFVA Hydraulic Supply Manifold K405, K406, K407

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					Root Fla	tical Initial w Size Not			-	
					Side Not D	etectable				
Component	Basic Part Number	Weld N	lumber Weld Type	Class	Access HC	F LCF	C	Comments		
MANIFOLD	RS007120	1-7	GTAW	1	х х					