SSME EA/CIL REDUNDANCY SCREEN

Component Group:

Ducts and Lines

CIL Item: Part Number:

N900-01 RS007287

Component:

GCV Control Line

FMEA Item: Failure Mode:

N900 Fails to contain helium. Prepared: Approved: D. Early T. Nguyen

Approval Date: Change #:

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Directive #:

CCBD ME3-01-5638

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Phase	Failure / Effect Description	Criticality Hazard Reference	
C 4.1	Helium leakage into aft compartment. HPV fails to actuate. Loss of Pogo shutdown charge during MECO, at zero G condition and minimum NPSP, will result in cavitation/overspeed of HPOTP and/or LPOTP. Loss of vehicle.	1 ME-C1A,C	
	Redundancy Screens: SINGLE POINT FAILURE: N/A		

SSME FMEA/CIL DESIGN

Component Group:

Ducts and Lines

CIL Item: Part Number: N900-01 RS007287

Component:

GCV Control Line

FMEA Item:

N900

Failure Mode:

Fails to contain helium.

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Design / Document Reference

FAILURE CAUSE: A: Parent material failure or weld failure.

THE LINE ASSEMBLY (1) IS MANUFACTURED UTILIZING 321 CRES TUBE AND INCONEL 625 BAR FOR ELBOW AND FLANGES. 321 CRES TUBING WAS SELECTED BECAUSE OF ITS STRENGTH, FABRICABILITY, GENERAL CORROSION RESISTANCE AND STRESS CORROSION RESISTANCE (2). INCONEL 625 WAS SELECTED FOR ITS WELDABILITY, FORMABILITY, RESISTANCE TO STRESS CORROSION CRACKING, AND CORROSION RESISTANCE (2). INCONEL 625 POSSESSES THE REQUIRED STRENGTH WITHOUT REQUIRING HEAT TREAT. FLANGE AND ELBOW SECTIONS INCORPORATE RADIUS JOINTS TO REDUCE STRESS CONCENTRATIONS. OFFSET LIMIT REQUIREMENTS ARE ESTABLISHED TO REDUCE STRESS CONCENTRATIONS AND IMPROVE WELD GEOMETRY. TUBING STOCK IS DRAWN TO MAINTAIN SURFACE REGULARITY, INSTALLATION IS CONTROLLED FOR ANGULARITY AND OFFSET PER SPECIFICATION REQUIREMENTS (3). MINIMUM FACTORS OF SAFETY FOR THE LINE MEET CEI REQUIREMENTS (4). HIGH AND LOW CYCLE FATIGUE LIFE MEET CEI REQUIREMENTS (5). THE LINE ASSEMBLY HAS COMPLETED PRESSURE CYCLING AND ULTIMATE PRESSURE DVS TESTING (6). THE LINE ASSEMBLY PARENT MATERIALS WERE CLEARED FOR FRACTURE MECHANICS/NDE FLAW GROWTH, SINCE THEY ARE NOT FRACTURE CRITICAL PARTS (7). TABLE N900 LISTS ALL THE FMEA/CIL WELDS AND IDENTIFIES THOSE WELDS IN WHICH THE CRITICAL INITIAL FLAW SIZE IS NOT DETECTABLE, AND THOSE WELDS IN WHICH THE ROOT SIDE IS NOT ACCESSIBLE FOR INSPECTION. THESE WELDS HAVE BEEN ASSESSED AS ACCEPTABLE FOR FLIGHT BY RISK ASSESSMENT (8).

(1) RS007287; (2) RSS-8582; (3) RA1102-006; (4) RSS-8586, CP320R0003B; (5) RL00532, CP320R0003B; (6) RSS-511-43; (7) NASA TASK 117; (8) RSS-8756

SSME FME **INSPECTION AND TEST**

Component Group:

Ducts and Lines

CIL Item: Part Number: N900-01 RS007287

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Failure Causes	Significant Characteristics	Inspection(s) / Test(s)	Document Reference
	GOX CONTROL LINE ASSEMBLY		RS007287
	MATERIAL INTEGRITY	MATERIAL INTEGRITY IS VERIFIED PER DRAWING REQUIREMENTS.	RS007287 RS007147 RS007237
		THE MACHINED DETAILS ARE PENETRANT INSPECTED PER SPECIFICATION REQUIREMENTS.	RA0115-116
	WELD INTEGRITY	ALL WELDS ARE INSPECTED TO DRAWING AND SPECIFICATION REQUIREMENTS PER WELD CLASS. INSPECTIONS INCLUDE: VISUAL, DIMENSIONAL, PENETRANT, RADIOGRAPHIC, ULTRASONIC, AND FILLER MATERIAL, AS APPLICABLE.	RL10011 RA0607-094 RA0115-116 RA0115-006 RA1115-001 RA0115-127
	ASSEMBLY INTEGRITY	THE LINE ASSEMBLY IS PROOF PRESSURE TESTED PER DRAWING REQUIREMENTS.	RS007287
		WELDS ARE PENETRANT INSPECTED AFTER PROOF PRESSURE TEST.	RA0115-116
	FLIGHT FLOW TESTING	THE EXTERNAL SURFACE IS VISUALLY INSPECTED PRIOR TO EACH LAUNCH. (LAST TEST)	OMRSD V41BU0.03

Failure History:

Comprehensive failure history data is maintained in the Problem Reporting database (PRAMS/PRACA)

Reference: NASA letter SA21/88/308 and Rocketdyne letter 88RC09761.

Operational Use: Not Applicable.

SSME FMEA/CIL **WELD JOINTS**

Component Group:

Ducts and Lines

CIL Item:

N900

Part Number:

RS007287

Component:

GCV Control Line

FMEA Item:

N900

Prepared:

D. Early

Approved: Approval Date:

T. Nguyen 7/25/00

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Critical Initial

Root

Flaw Size Not

Side Not

Detectable

Component

Basic Part Number

Weld Number Weld Type

Class

Access HCF LCF

LINE

RS007287

1-3

GTAW

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Comments