

CIL
EMU CRITICAL ITEMS LIST

12/24/91 SUPERSEDES 08/31/90

ANALYST:

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NAME P/N QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
CAUTION AND WARNING SYSTEM, EYEN 150 ----- 6N785970-13 (1) *GENERIC FAILURE MODE. COMMON FAILURE MODES 150FH05. (1319-1) ----- 2	2/100	150FH12: Electrical short, 5.6 or 14.2 volt power leads. CAUSE: Contamination on the electrical connector faulty leads or electronic component failure.	EMU ITEM: Electrical short to ground on 5.6 volt or 14.2 volt line from DCN DC/DC converter. GPE INTERFACES: Increase in battery power consumption. The current is limited in the DCN DC/DC converter 1.8 +/- 0.25 amps. Shutdown of the DC/DC converter. Loss of CWS, tones and DCN display. MISSION: None for single failure. Terminate EVA with loss of DCN display, CWS and ability to monitor the operational integrity of the EMU. Loss of use of one EMU. CREW/VEHICLE: None for single failure. Possible loss of crewman with loss of EDC, oxygen or low vent flow.	A. Design - Established reliability capacitors and resistors are qualified to applicable military standards and thermal shocked per Condition B Test Method 107 of MIL-STD-202. Microcircuits are qualified to the requirements of MIL-H-30570 and receive the burn-in of Class B parts per Method 5004 of MIL-S10-003. Transistors, diodes are qualified to the requirements of MIL-S-18500 and receive the burn-in of JANTRY level parts per the applicable methods, 1038, 1039, 1040 of MIL-STD-750. The electronic components are operating within the power dissipating requirements of 5VHS2804. The printed circuit (PC) boards are fiberglass/epoxy per MIL-P-13949 type G7 and manufactured in accordance with SN-P-0006. Parts mounting and soldering is per MSFC-STD-156 and 5VHS388.4 (3A-1). The CWS is a mother/daughter board assembly. The daughter boards are held in place by metal card guides which also provide thermal transfer from the boards to the CWS case. The top cover of the CWS exerts a downward force on the daughter boards to keep them properly seated in the mother board connectors. Flex Tape (Kapton Insulated, flexible flat conductor) instead of conventional Teflon coated wires is used to provide connections between the mother board and the external connectors. This prevents pinching of the conductor during item assembly. The PC board assemblies are conformal coated per MIL-A-46146 (Dow Corning RTV 3140) for environmental and humidity protection. Electrical connectors are environmentally sealed to prevent damage due to contamination and humidity. B. Test - Component Acceptance Test - Full functioning of the CWS is verified during Item AIP Tests include continuity, logic flow, x-state sequencing, fault simulation, verification of status and fault messages, warning and alert tones activation, and BITE activation. These tests are conducted upon completion of random vibration testing (6.0 gms). FOA Test - The above electrical tests are repeated during PLSS FOA to verify CWS operation. The CWS is also operational during

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	2/100	150FH12:		<p>other PLSS PBA Electrical tests such as sensor accuracy checks, Item 123 Fan Operation, Item 174 RTDS Checkout, and Solenoid Valve Actuation.</p> <p>Certification Test - The item completed the 15 year structural vibration and shock certification requirement during 10/85. EC's 42806-264 (add jumper wires, add diode CR229, change resistor R301), 42806-345-3 (eliminate interferences with PLSS), 42806-718 (overstressed resistor R305 due to an improper interface circuit in the data delta logic, software change, diode CR201 rewiring), 42806-962 and 42806-962-1 (transistor Q201 lead stress relief) have been incorporated since this configuration was tested.</p> <p>C. Inspection - Each circuit board, the item tape, and connectors are inspected for damage and contamination prior to being placed into finished stores. The CMS assembly is inspected internally and externally for damage and contamination during item assembly and externally during ATP. All soldering is inspected by HB QA and DCAS QA per MMS1900.4 (3A-1).</p> <p>D. Failure History - None.</p> <p>E. Ground Turnaround - Tested per FEMU-B-801, DCN display verification.</p> <p>F. Operational Use - Crew Response - PreEVA: Trouble shoot problem, if no success, consider third EMU if available. EMU no go for EVA. EVA: When loss of CMS tones and displays detected, terminate EVA. Training - Standard EMU training covers this failure mode. Operational Considerations - flight rules define an operational CMS as at least able to monitor a valid status list. EVA Checklist procedures verify hardware integrity and systems operational status prior to EVA. Real Time Data</p>

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	2/1RD	15DIM12:		System allows ground monitoring of EMU systems.