

01/02/90 SUPERSEDES / /

ANALYST:

NAME P/N REV	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
COOLING CONTROL VALVE, ITEM 321 ----- SV789693-1 (1)	2/18	321FHD4A1 External water leakage. CAUSE: Seal failure. Spool seat failure (rotating seal backed up by a redundant chevron face seal, housing static seal failure.	ERD ITEM: Water leakage to ambient. CPE INTERFACE: Depletion of the water reservoir, reduction in heat rejection. MISSION: Terminate EVA when the water supply drops below CMS limits. CREW/VEHICLE: None for single failure. Possible loss of crewman with loss of SOP.	A. Design - Two Transport in-line filters (146 micron, SCU BCM side and 20 micron 1-141 Gas Trap) protect the valve from contamination. In addition, supplemental water from the condensate circuit is filtered by a 20 micron filter contained in the 3-127. The static seal and rotating seal are silicon and Viton, respectively, and their design configuration, dimensions and rigidity of assembly provide squeeze under all loading conditions. B. Test - Component Acceptance: An external leakage test is performed per AT-3-321-2 in which the valve is pressurized with nitrogen to 61.7 to 63.7 psia. The valve is then submerged in water for a 5 minute minimum test period. No leakage bubbles are allowed. PDA: An external leakage test is performed per SEMU-60-015. The valve is pressurized with water to 15.0-20.0 psig and observed for evidence of external leakage for a 5 minute minimum test period. No visible leakage is allowed. Certification: The item completed the 15 year structural vibration and shock certification requirement during 10/83. The item completed 10,000 cold-hot-cold cycles during 7/85 which fulfills the cycle certification requirement of 4,024. Engineering changes 42806-329 (facilitated valve acceptance at DCM level by providing consistency between component spec. and S/AD 11) and 42806-515 (clarified Flow Requirement) have been incorporated and certified by analysis/similarity since this configuration was tested. C. Inspection - Spool seal failure (rotating seal), housing static seal failure. O-ring grooves are 100% inspected per drawing dimensions and surface finish. O-rings are inspected for surface characteristics per SVHS 3432: 100% for Class I & II and at least 1.5 AQL for Class III.

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 EMU CRITICAL ITEMS LIST

12/24/91 SUPERSEDES 01/02/90

Page: 2
 Date: 12/03/91

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	2/1R	321#H04A:		<p>D. Failure History - None.</p> <p>E. Ground turnaround - Tested per FEMU-R-001, Water Servicing, Leakage and Gas Removal.</p> <p>F. Operational Use - Crew Response - Pre/PostEVA: Troubleshoot problem, if no success, consider third EMU if available. Otherwise, terminate EVA operations. EVAs when CUS data confirms loss of feedwater and cooling is insufficient, terminate EVA. Consider vacuum water recharge to recover EMU operation. Special Training - Standard training covers this failure mode. Operational Considerations - EVA checklist procedures verify hardware integrity and systems operational status prior to EVA. Flight rules define go/no go criteria related to EMU thermal control. Real Time data system allows ground monitoring of EMU systems.</p>