FMEA

REV.

2

REF.

3610

MAME, QTY, &

DRAWING REF.

MOTOR MODULE

ASSEMBLY

QTY-1 P/H

51140E1473

51140E2203

DESIGNATION

PROJECT: SRMS SREE1: \_\_\_1 ASS'Y P/N: 51140E1470-14-3 ASS'Y NOMENCLATURE: END EFFECTOR RATIONALE FOR ACCEPTANCE FAILURE EFFECT HDWR / FUNC. FAILURE MODE 2/1R AND SCREENS: A-PASS, B-PASS, C-PASS CRITICALITY END ITEM CAUSE DESIGN FEATURES MOTOR WILL NOT MODE: DRIVE. LOSS OF THEREFORE THE END EFFECTOR MOTOR IS A MAJOR BOUGHT-OUT-PART WHICH IS SUPPLIED BY HONEYWELL SPERRY CORPORATION AND MEETS OR MOTOR DRIVE. SYSTEM WILL BE INOPERATIVE IN CAUSE(S): EXCEEDS THE REQUIREMENTS OF SPECIFICATION SPAR-SG.446 AUTO AND MANUAL (1) MOTOR FOR THE P/N 511400575-1 AND SPAR-SG.1092 FOR P/N 511400575-3 END EFFECTOR WINDING. HODES. SHORT LIMP ARM DURING CIRCUIT. AUTO CAP. SEQ. (2) LOSS OF THE MOTOR COMPRISES OF: UNTIL EE MODE MOTOR DRIVE SW TO OFF. A MULTIPOLE ROTOR BUILT WITH "RARE EARTH" PERMANENT MAGNETS. FROM EEEU. **HORST CASE** A WOUND STATOR, CONSISTING OF 24 COILS WOUND IN GROUPS OF 8. THE 3 GROUPS ARE SYMMETRICALLY ARRANGED AND INSERTED IN 24 COMMUTATION FAILURE/ UNEXPECTED RADIAL SLOTS IN A LAMINATED STEEL CORE. THE ENDS OF THE 3 COIL MECHANICAL. PAYLOAD MOTION. GROUPS ARE JOINED AND CONNECTED TO TEFLON INSULATED LEAD INCOMPLETE ELECTRICAL WIRES TO FORM THE CONVENTIONAL "DELTA" CONFIGURATION. OR LOSS OF CAPTURE/RELEASE SEQUENCE. THE WINDING FEATURES THAT HELP PREVENT SHORT OR OPEN CIRCUITS PLATING FROM UNABLE TO CODE WHEEL. RELEASE (4) SEIZURE PAYLOAD. CREW OF MOTOR OR - INSULATION IS TO CLASS 185 (H) ACTION REQ. CLUTCH INPUT - WIRE USED IN HEAVY ML MAGNET WIRE. - COILS ARE BAKED TO STRESS RELIEVE COPPER AND INSULATION. BEARINGS. REDUNDANT PATHS (5) INPUT - SLOTS HAVE POLYMIDE LINER. REMAINING LEAD TO END WINDINGS ARE ENCLOSED IN FIBREGLASS COVERS. MOTOR OPEN WINDING IS VACUUM IMPREGNATED USING 100% SOLID EPOKY, THIS BACKUP EE CIRCUIT. IMPARTS GOOD THERMAL AND MECHANICAL PERFORMANCE. RELEASE. (6) HOTOR SHAFT THE DESIGN UTILIZES PROVEN CIRCUIT TECHNIQUES AND IS BREAKS. IMPLEMENTED USING CHOS LOGIC DEVICES. CHOS DEVICES OPERATE AT LOW POWER AND HENCE DO NOT EXPERIENCE SIGNIFICANT OPERATING STRESSES. THE TECHNOLOGY IS MATURE, AND DEVICE RELIABILITY HISTORY IS WELL DOCUMENTED. ALL STRESSES ARE ADDITIONALLY REDUCED BY DERATING THE APPROPRIATE PARAMETERS IN ACCORDANCE WITH SPAR-RMS-PA.003. SPEC!AL HANDLING PRECAUTIONS ARE USED AT ALL STAGES OF MANUFACTURE TO PRECLUDE DAMAGE/STRESS DUE TO ELECTROSTATIC DISCHARGE. CONNECTOR USED ARE TO GSFC SPECIFICATION S.311.P.4/9.

CONTACTS USED ARE TO GSF SPEC.S.311.P.4/9.

CRIMPING IS CONTROLLED TO SPAR PPS 9:17 WHICH EMBODIES

MEETS OR EXCEEDS THE REQUIREMENTS OF SPAR-SG.454.

THE END EFFECTOR CONMUTATION SCANNER ASSEMBLY (CSA) IS A MAJOR BOUGHT-OUT-PART WHICH IS SUPPLIED BY BEL MOTION SYSTEMS AND

TIES-13 TYPE LED IS PROCURED TO A JANTXV-EQUIVALENT LEVEL SPECIFICATION, THEN RESCREENED AGAIN TO INSURE THAT INFANT FAILURES ARE REMOVED. THE LED IS OPERATING AT A CURRENT STRESS RATIO OF 50/300, OR 0.2, WHICH IS A VERY LOW LEVEL. THIS MINIMIZES DEGRADATION OF THE LIGHT OUTPUT WITH TIME. THIS TYPE OF DEVICE HAS BEEN CHARACTERIZED WITH RESPECT TO RADIATION,

SYSTEM: MECHANICAL ARM SUBSYSTEM

DATE: 24 JUL 91 CIL REV: 2

MSC-SPEC-Q-1A.

PREPARED BY:

MFWG

ITICAL IT	EMB LIST	PR AS	OJECT: SRMS S'Y NOMENCLATURE: <u>En</u>	SYSTEM: MECHANICAL ARM SUBSYSTEM ASS'Y P/N: 51140E1470-18-3 SHEET:
FMEA FMEA REF. REV.	MAME, QTY, & DRAWING REF. DESIGNATION	FAILURE MODE AND CAUSE	FAILURE EFFECT ON END ITEM	HDMR / FUNC. RATIONALE FOR ACCEPTANCE 2/1R CRITICALITY SCREENS: A-PASS, B-PASS, C-PASS
3610 2	MOTOR MODULE ASSEMBLY QTY-1 P/N 51140E1473 OR 51140E2203	MODE: LOSS OF MOTOR DRIVE.  CAUSE(S): (1) MOTOR WINDING SHORT CIRCUIT. (2) LOSS OF MOTOR DRIVE FROM EEEU. (3) COMMUTATION FAILURE/ MECHANICAL OR LOSS OF PLATING FROM CODE WHEEL. (4) SEIZURE OF MOTOR OR CLUTCH INPUT BEARINGS. (5) IMPUT LEAD TO MOTOR OPEN CIRCUIT. (6) MOTOR SHAFT BREAKS.	MOTOR WILL NOT DRIVE. THEREFORE, SYSTEM WILL BE INOPERATIVE IN AUTO AND MANUAL END EFFECTOR MODES. LIMP ARM DURING AUTO CAP. SEQ. UNTIL EE MODE SW TO OFF. WORST CASE UNEXPECTED PAYLOAD MOTIOM. INCOMPLETE CAPTURE/RELEASE SEQUENCE. UNABLE TO RELEASE PAYLOAD. CREW ACTION REQ. REDUNDANT PATHS REMAINING BACKUP EE RELEASE.	AND THE EXPECTED DEGRADATION IS VERY NOMINAL (2-5%, FOR 7 YEARS IN GEOSYNCHRONOUS ORBIT).  THE LED IS ASSEMBLED INTO AN ALUMINUM MOUNTING RING, USING A THERMALLY COMDUCTIVE EPOXY.  THE FIBER OPTICS USED ON THE RMS COMM SCANNERS ARE A CUSTOM DESIGN, MANUFACTURED BY GALILEO ELECTRO-OPTICS CORPORATION.  THE FIBER OPTIC BUNDLES ARE SECURED AT EACH END BY METAL RINGS AND EPOXY. THE BUNDLE LENGTHS ARE SUPPORTED BY A FLEXIBLE MOVEM GLASS TUBE AND A STAINLESS STEEL SPRING. STRESS RELIEF ARE USED AT THE ANCHOR POINTS.  ALL SEE PARTS ARE PROCURED TO MILITARY SPECIFICATIONS OR EQUIVALENT. THE CIRCUITS EMBRODY THE USE OF MHSS300.4 (3A) SOLDERING, WITH NO PLATED-THRU HOLES (2 WHRES ARE USED WHERE NECESSARY) AND ALL LAP SOLDER JOINTS. THE EMI FILTER IS PURCHASED TO AN SCO (905-15815) WHICH INCOMPORATES RESCREENING INCLUDING THERMAL SHOCK, BURN-IN, AND HERMETICITY TESTING, AS WELL AS X-RAY OF ALL UNITS.  CERAMIC CAPACITORS ARE USED THROUGHOUT. THE BUS CAPACITORS ARE SLEVEL MS9014.  THE CURRENT LIMIT RESISTOR (LED 50MA) IS A TMO WAIT RATING RUMBOS TYPE DEVICE, OPERATING AT A STRESS LEVEL OF LESS THAN 0.1 VG GIVE A VERY LOW PROBABILITY OF FAILURE.  THE CURRENT CONFIGURATION PHOTOCELL IS ASSEMBLED AT BET USING SPAR-APPROVED PROCEDURES. IT IS SCREENED AND GUALIFIED PER A BEI SCD (905-16816) TO STRESS LEVELS FAR IN EXCESS OF MISSION LIMITS.  CODE WHEELS ARE MANUFACTURED PER BEI PROCEDURE 905A12224. THE BASE METAL WHEEL IS CHROME-PLATED BY AM OUTSIDE VENDOR, AND RETURNED TO BEI. PHOTO-RESIST IS USED TO MASK AREAS WHICH WALL REMAIN BLACK, WHILE AREAS WHICH ARE TO BE REFLECTIVE ARE STRIPPED OF BLACK CHROME TO EXPOSE THE GOLD SUFFACE. THIS PROCESS INSURES GOOD ADHESION.  THE END EFFECTOR PRIME AND BOCK-UP RELEASE CLUTCH DESIGNS UTILIZE THREE BEARINGS, TWO OF WHICH ARE IDENTICAL. THE BEARINGS ARE PERMANENTLY LUBRICATED WITH MET LUBRICANT. THE TWO IOENTICAL BEARINGS ARE SCALED WITH TEFLON COATED FIREGELASS SEALS, BOTH SIDES, TO PREVENT THE INGRESS OF DEBRIS.  THE GREASE IS APPLIED IN PRECISE QUANTITY TO EACH BEARING.  THE LIFE OF THE BE

SUPERCEDING DATE: 06 OCT 87

SYSTEM: MECHANICAL ARM SUBSYSTEM
ASS'Y P/N: 51140E1470-18-3 SHEET: 3 PROJECT: SRMS ASS'Y NOMENCLATURE: END EFFECTOR

AND THE PROPERTY OF THE PROPER				W9:	S'T NUMENCLATURE: EI	
CASSEMBLY OFFI-1 P/M STI-MCE1473 OFFI-1 P/M S		FMEA REV.	DRAWING REF.	AND	ON	
DATE: 24 JUL 91 CIL REV:	3610	2	ASSEMBLY QTY-1 P/N 51140E1473	LOSS OF MOTOR DRIVE.  CAUSE(S): (1) MOTOR WINDING SHORT CIRCUIT. (2) LOSS OF MOTOR DRIVE FROM EEEU. (3) COMMUTATION FAILURE/ MECHANICAL OR LOSS OF PLATING FROM CODE WHEEL. (4) SEIZURE OF MOTOR OR CLUTCH INPUT BEARINGS. (5) INPUT LEAD TO MOTOR OPEN CIRCUIT. (6) MOTOR SHAFT BREAKS.	DRIVE. THEREFORE SYSTEM WILL BE INOPERATIVE IN AUTO AND MANUAL END EFFECTOR MODES. LIMP ARM DURING AUTO CAP. SEG. UNTIL EE MODE SW TO OFF.  WORST CASE  UNEXPECTED PAYLOAD MOTION. INCOMPLETE CAPTURE/RELEASE SEQUENCE. UNABLE TO RELEASE PAYLOAD. CREW ACTION REQ.  REDUNDANT PATHS REMAINING  BACKUP EE RELEASE.	MINISTOD. 4 (3A) REQUIREMENTS. THE CONNECTIONS ARE SUPPORTED MITHIN THE FIBREGLASS END WINDING COVER.  - STRUCTURAL CONSIDERATIONS  THE MINIMUM MARGINS OF SAFETY FOR THE DRIVE ARE:  SHAFT E1593

PREPARED BY:

HFWG

SYSTEM: MECHANICAL ARM SUBSYSTEM SHEET: 4 PROJECT: SRMS CRITICAL ITEMS LIST ASS'Y P/N: 51140E1470-14-3 ASS'Y NOMENCLATURE: END EFFECTOR RATIONALE FOR ACCEPTANCE HDWR / FUNC. 2/1R **FAILURE EFFECT** FAILURE MODE NAME, QTY, & DRAWING REF. FMEA **FMEA** ON AND SCREENS: A-PASS, B-PASS, C-PASS REV. REF. CRITICALITY END LTEM CAUSE DESIGNATION ACCEPTANCE TESTS MOTOR WILL NOT MOTOR MODULE HODE: 2 3610 DRIVE. THE EE ASSEMBLY IS TESTED TO THE FOLLOWING ACCEPTANCE LOSS OF ASSEMBLY QTY-1 P/W THEREFORE HOTOR DRIVE. ENVIRONMENTS: SYSTEM WILL BE 51140E1473 INOPERATIVE IN O VIBRATION: LEVEL AND DURATION - REFERENCE TABLE 7 CAUSE(S): AUTO AND MANUAL 51140E2203 (1) HOTOR END EFFECTOR O THERMAL VACUUM: +70 DEGREES C TO -25 DEGREES C (1 1/2 WINDING MODES. SHORT CYCLES) 1 X 10\*\*6 TORR LIMP ARM DURING CIRCUIT. AUTO CAP. SEQ. UNTIL EE MODE THE EE ASSEMBLY IS FURTHER TESTED IN THE IN THE RMS SYSTEM (2) LOSS OF TEST (TP518 RMS STRONGBACK AND TP552 FLAT FLOOR TESTS) WHICH HOTOR DRIVE SW TO OFF. FROM EEEU. VERIFIES THE ABSENCE OF THE FAILURE MODE. COMMUTATION **MORST CASE** QUALIFICATION TESTS FAILURE/ UNEXPECTED THE EE ASSEMBLY QUALIFICATION TESTING CONSISTED OF THE MECHANI CAL PAYLOAD MOTION. FOLLOWING ENVIRONMENTS: INCOMPLETE ELECTRICAL CAPTURE/RELEASE O VIBRATION: LEVEL AND DURATION - REFERENCE TABLE 7 OR LOSS OF PLATING FROM SEQUENCE. UNABLE TO 20G/11 MS - 3 AXES (6 DIRECTIONS) CODE WHEEL. O SHOCK: RELEASE (4) SEIZURE PAYLOAD. CREW O THERMAL VACUUM: +81 DEGREES C TO -36 DEGREES C (6 CYCLES)
1 X 10\*\*6 TORR OF MOTOR OR ACTION REQ. CLUTCH IMPUT BEARINGS. REDUNDANT PATHS 95% RH (65 DEGREES C MAINTAINED FOR 6 HRS) (5) IMPUT O HUMIDITY: REMAINING (65 DEGREES C TO 30 DEGREES C IN 16 HRS). 10 CYCLES 240 HRS. LEAD TO HOTOR OPEN BACKUP EE CIRCUIT. MIL-STD-461A AS MODIFIED BY SL-E-0002 (TEST RELEASE. (6) MOTOR O EMC: CEO1, CEO3, CSO1, CSO2, CSO6, REO2 (N/8)) SHAFT BREAKS. O STRUCTURAL STIFFNESS AND LOAD TEST FLIGHT CHECKOUT PDRS OPS CHECKLIST (ALL VEHICLES) JSC 16987

RMS/MECH - 4

APPROVED BY:

SUPERCEDING DATE: 06 OCT 87

DATE: 24 JUL 91

CIL REV: \_2

PROJECT: SRMS
ASS'Y NOMENCLATURE: END EFFECTOR

SYSTEM: MECHANICAL ARM SUBSYSTEM
ASS'Y P/N: 51140E1470-18-3 SHEET: 5

	T		***************************************	PATRIME PERCE	HOWR / FUNC. RATIONALE FOR ACCEPTANCE
FMEA REF.	FMEA REV.	NAME, QTY, & Drawing Ref.	FAILURE MODE AND	FAILURE EFFECT	2/1R
		DESIGNATION	CAUSE	END LTEM	CRITICALITY SCREENS: A-PASS, B-PASS, C-PASS
3610	2	HOTOR MODULE ASSEMBLY GTY-1 P/M 51140E1473 OR 51140E2203	MODE: LOSS OF MOTOR DRIVE.  CAUSE(S): (1) MOTOR WINDING SHORT CIRCUIT. (2) LOSS OF MOTOR DRIVE FROM EEEU. (3) COMMUTATION FAILURE/ MECHANICAL OR COS OF PLATING FROM CODE WHEEL. (4) SEIZURE OF MOTOR OR CLUTCH INPUT BEARINGS. (5) INPUT LEAD TO MOTOR OPEN CIRCUIT. (6) MOTOR SHAFT BREAKS.	MOTOR WILL NOT DRIVE. THEREFORE, SYSTEM WILL BE INOPERATIVE IN AUTO AND MANUAL END EFFECTOR MODES. LIMP ARM DURING AUTO CAP. SEQ. UNTIL EE MODE SW TO OFF. WORST CASE UNEXPECTED PAYLOAD MOTION. INCOMPLETE CAPTURE/RELEASE SEQUENCE. UNABLE TO RELEASE PAYLOAD. CREW ACTION REQ. REDUNDANT PATHS REMAINING BACKUP EE RELEASE.	UNITS ARE MANUFACTURED UNDER DOCUMENTED QUALITY CONTROLS. THESE CONTROLS ARE EXERCISED THROUGHOUT DESIGN PROCUREMENT, PLANNING, RECEIVING, PROCESSING, FABRICATION, ASSEMBLY, TESTING AND SHIPPING OF THE UNITS. MANDATORY INSPECTION POINTS ARE EMPLOYED AT VARIOUS STACES OF FABRICATION ASSEMBLY AND TEST. GOVERNMENT SOURCE INSPECTION IS INVOKED AT VARIOUS CONTROL LEVELS.  WIRE IS PROCURED TO SPECIFICATION MIL-W-22759 OR MIL-W-81381 AND INSPECTED AND TESTED TO MASA JSCHOOLD STANDARD NUMBER 95A.  RECEIVING INSPECTION VERIFIES THAT THE HARDWARE RECEIVED IS AS IDENTIFIED IN THE PROCUMENENT DOCUMENTS, THAT NO DAMAGE HAS OCCURRED DURING SHIPPING AND THAT APPROPRIATE DATA MAS BEEN RECEIVED WHICH PROVIDES ADEQUATE TRACEABILITY INFORMATION AND IDENTIFIES ACCEPTABLE PARTS.  PARTS ARE INSPECTED THROUGHOUT MANUFACTURE AND ASSEMBLY AS APPROPRIATE TO THE MANUFACTURING STAGE COMPLETED. THESE INSPECTIONS INCLUDE, MAGNET WIRE IS PROCURED TO MIL-W-583 AND CHECKED AT INCOMING INSPECTION PER FEDERAL STANDARD J-W-1177 WHICH INCLUDES DIELECTIC, PIN HOLES, BUBBLES, BLISTERS, AND CRACKS IN THE INSULATION.  ALL SOLDERING IS ACCOMPLISHED BY OPERATORS, HIMO ARE TRAINED AND CERTIFIED TO NASA MHB5300.4(3A) STANDARD, AS MODIFIED BY JSC OBBOODA.  BEARINGS RECEIVE DIMENSIONAL INSPECTION AT THE SUPPLIER AND VERTICATION BY SPAR RECEIVING INSPECTION TRACKS AND INNER/JUIER RACE DIAMETERS. AFTER ASSEMBLY PRIOR TO LUBBICATION, ROBIAL CLEARANCE MEASUREMENTS ARE TAKEN. FOLLOWING LUBRICATION, ROBIAL CHEARNE MEASUREMENTS ARE TAKEN. FOLLOWING LUBRICATION, ROBIAL CHEARNE MEASUREMENTS. CHEARINGS PORY LUBB EBRAINGS SPECIALIZED BEARING INSPECTION EQUIPMENT AT SPAR IS USED TO VERIFY QUALITY AND STICTION LEVELS THROUGH STRIP CHARNE MEASUREMENTS. GOVERNENT SARE TAKEN. FOLLOWING LUBRICATION, REPORTED TO VERIFY THAT INVOLUTE FORM, PITCH CIRCLE CONCENTRICITY AND PITCH OTAMETER ARE TO DRAMING REQUIREMENTS. THIS INSPECTION AND DEALITY THAT COMPOSITE ERROR GEAR CHECKER IS USED TO VERIFY THAT INVOLUTE FORM, PITCH CIRCLE CONCENTRICITY AND PITCH OTAMETER ARE TO DRAMING REGU
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SYSTEM: MECHANICAL ARM SUBSYSTEM SHEET: 6 PROJECT: SRMS
ASS'Y NOMENCLATURE: END EFFECTOR

OFF REV. DR	ME, QTY, &	FAILURE MODE	FAILURE EFFECT	HDWR / FUNC. RATIONALE FOR ACCEPTANCE 2/1R CRITICALITY SCREENS: A-PASS, B-PASS, C-PASS
3610 2 HO AS GT 51	SIGNATION TOR MODULE SEMBLY Y-1 P/N 140E1473	MODE: LOSS OF MOTOR DRIVE.  CAUSE(S): (1) MOTOR WINDING SHORT CIRCUIT. (2) LOSS OF MOTOR DRIVE FROM EEEU. (3) COMMUTATION FAILURE/ MECHANICAL OR CLUTCH INPUT BEARINGS. (5) IMPUT LEAD TO HOTOR OPEN CIRCUIT. (6) MOTOR SHAFT BREAKS.	MOTOR WILL MOT DRIVE. THEREFORE, SYSTEM WILL BE INOPERATIVE IN AUTO AND MANUAL END EFFECTOR MODES. LIMP ARM DURING AUTO CAP. SEQ. UNTIL EE MODE SW TO OFF. WORST CASE UNEXPECTED PAYLOAD MOTION. INCOMPLETE CAPTURE/RELEASE SEQUENCE. UNABLE TO RELEASE PAYLOAD. CREW ACTION REQ. REDUNDANT PATHS REMAINING BACKUP EE RELEASE.	SHAFTS ARE DIMENSIONAL INSPECTED TO DRAWING REQUIREMENTS THROUGHOUT THE MANUFACTURING STAGES. FOLLOWING HEAT TREATMENT THE SHAFTS ARE SUBJECTED TO MAGNETIC PARTICLE INSPECTION FOR CRACKS.  INSPECTION VERIFIES THAT KITTED PARTS ARE CORRECT PRIOR TO ASSEMBLY AND TRACEABILITY INFORMATION RECORDED.  INSPECTION TO DRAWING IS CONDUCTED THROUGHOUT THE ASSEMBLY PROCESS, INCLUDING INSPECTION OF LOCKING, WITNESSING OF TORQUING AND APPLICATION OF TORQUE STRIPING.  MOTOR MODULES ARE TESTED TO THE REQUIREMENTS OF SPAR-TM. 1624 WHICH INCLUDES, CONTINUITY AND ISOLATION CHECKS, STICTION, COMPUTATOR TIMING, AMBIENT AND THERMAL TESTING.  (SPAR/GOVERNMENT REP MANDATORY INSPECTION POINT).  INTEGRATION OF MOTOR MODULE TO END EFFECTOR LRU- INSPECTIONS INCLUDE GROUNDING CHECKS, CONNECTORS FOR BENT OR PUSHBACK CONTACTS, INCONNECT WIRING ETC.  PRE-ACCEPTANCE TEST INSPECTION, WHICH INCLUDES AN AUDIT OF LOWER TIER INSPECTION COMPLETION, AS BUILT CONFIGURATION VERIFICATION TO AS DESIGN ETC., (MANDATORY INSPECTION POINT).  A TEST READINESS REVIEW (TRY) WHICH INCLUDES VERIFICATION OF TEST PERSONNEL. TEST DOCUMENTS, TEST GOUIPMENT CALIBRATION/ VALIDATION STATUS AND MARDWARE CONFIGURATION IS CONVENED BY QUALITY ASSURANCE IN COMJUNCTION WITH HEGINERING, RELIABILITY, COMPIGURATION CONTROL, SUPPLIER AS APPLICABLE, AND THE GOVERNENT REPRESENTATIVE, PRIOR TO THE START OF ANY FORMAL TESTING (ACCEPTANCE OR QUALIFICATION).  ACCEPTANCE TESTING (ATP) INCLUDES, AMBIENT, VIBRATION AND THERMAL-VAC TESTING, (SPAR/GOVERNMENT REP MANDATORY INSPECTIONS ARE PERFORMED AT EACH PHASE OF INTEGRATION WHICH INCLUDES GROUNDING CHECKS, THRU VIRING CHECKS, WIRING ROUTING, INTERFACE CONNECTORS FOR BENT OR PUSH BACK CONTACTS ETC.  SRMS SYSTEMS INTEGRATION, THE INJEGRATION OF MECHANICAL ARM SUBASSEMBLIES AND THE FLIGHT CABIN EQUIPMENT TO FORM THE SRMS. INSPECTIONS ARE PERFORMED AT EACH PHASE OF INTEGRATION WHICH INTERFACE CONNECTORS FOR BENT OR PUSH BACK CONTACTS ETC.  SRMS SYSTEMS TESTING - STRONGBACK AND FLAT FLOOR AMBIENT PERFORMANCE TEST. (SPAR/GOVERNMENT REP

FMEA

REF.

3610

SYSTEM: MECHANICAL ARM SUBSYSTEM
ASS'Y P/N: 51140E1470-18-3 SHEET: 7 PROJECT: SRMS
ASS'Y NOMENCLATURE: END EFFECTOR NAME, OTY, & DRAWING REF. FAILURE EFFECT HDWR / FUNC. RATIONALE FOR ACCEPTANCE FAILURE MODE **FMEA** 2/1R REV. AND CRITICALITY SCREENS: A-PASS, B-PASS, C-PASS CAUSE END ITEM DESIGNATION FAILURE HISTORY NOTOR WILL NOT 2 HOTOR HODULE MODE: LOSS OF DRIVE. ASSEMBLY QTY-1 P/N THEREFORE NOTOR DRIVE. SYSTEM WILL BE INOPERATIVE IN THERE HAVE BEEN NO FAILURES ASSOCIATED WITH THIS FAILURE 51140E1473 MODE ON THE SRMS PROGRAM. CAUSE(S): (1) MOTOR AUTO AND MANUAL 51140E2203 WINDING END EFFECTOR HODES. LIMP ARM DURING SHORT CIRCUIT. AUTO CAP. SEQ. (2) LOSS OF UNTIL EE MODE **HOTOR DRIVE** FROM EEEU. SW TO OFF. COMMUTATION WORST CASE FAILURE/ UNEXPECTED MECHANICAL PAYLOAD MOTION. INCOMPLETE ELECTRICAL CAPTURE/RELEASE SEQUENCE. OR LOSS OF PLATING FROM UNABLE TO CODE WHEEL. (4) SEIZURE RELEASE PAYLOAD. CREW OF MOTOR OR CLUTCH INPUT ACTION REQ. BEARINGS. (5) INPUT REDUNDANT PATHS REMAINING LEAD TO MOTOR OPEN CIRCUIT. BACKUP EE (6) MOTOR RELEASE. BREAKS.

DATE: 24 JUL 91 CIL REV: 2 APPROVED BY: SUPERCEDING DATE: 06 OCT 87 PREPARED BY: MFWG

MFMG

PREPARED BY:

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CIL REV: 2

DATE: 24 JUL 91

SUPERCEDING DATE: 06 OCT 87