

12/24/94 SUPERSEDES 12/24/92

ANALYST:

NAME P/N QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
POWER MODE SELECTOR SWITCH, ITEM 364 ----- SV778506-4 (1)	Z/2	364FM11: Electrical open at Power Switch position terminal. SDU (T4). CAUSE: Cold solder joint, severed lead wire, contamination on contact, broken contact.	END ITEM: Loss of SDU power connection to BCM and fan. GFE INTERFACE: Cannot power motor and OCM/DMS from vehicle power. MISSION: Terminate EVA. Loss of OCM and fan operation during EVA when in vehicle power mode. CREW/VEHICLE: None.	A. Design - Each of the three switches is sealed in a dry nitrogen filled, hermetically sealed case. The switches are per MIL-S-8805/46 with the 10 amp contacts silver plated. Microswitch contacts are rated for 10 amps. Actual current draw is 3.8 amps. The external solder terminals are designed to withstand an axial pull of 8 lbs without degradation. Switches are nickel silver to prevent oxidation of contacts. Microswitch actuator overtravel is adjusted to .007 inch minimum to ensure the common contact arm rotates completely over to the normally open contact. B. Test - Component Acceptance Test - Switch operation and continuity are verified during vendor acceptance tests. The switch is also subjected to 500 run-in cycles and an axial pull test on the handle to verify that it will not come loose during normal use. In-Process Test - Operation and integrity of the switch are verified during four separate in-process tests during initial item 350 assembly. These tests include continuity and output voltage. The switch is cycled during these tests. PDA Test - The switch is subjected to Acceptance/PDA testing as part of item 350. Tests include continuity, operating torque, vibration, thermal cycling, and thermal vacuum. The switch is also cycled during item 350 Acceptance/PDA electrical functional tests. Certification Test - The item completed the 15 year structural vibration and shock certification requirement during 10/85. The item completed 5,464 inductive and 8,536 resistive cycles during 1/81 which satisfied the cycle certification requirement of 5,464 and 8,536 respectively. Class I engineering change 42806-306 (Toggle Handle Pull Test) has been incorporated since this configuration was certified.

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	2/2	364PR11:		<p>C. Inspection - To preclude failure due to internal contamination, the switches are assembled by the vendor in an environmentally controlled room. Assembly and processing is per MIL-S-8805/46. The switches receive in-process cycling and leak checks. The entire item 364 is x-ray inspected for acceptability of brazing. The solder terminals on the switch are visually checked as part of source inspection for the part. The terminals are also inspected after lead wires are soldered on during OCM assembly. Solder joints are inspected per NH85300.4 (3A-1).</p> <p>D. Failure History - None.</p> <p>E. Ground Turnaround - Tested per FEMU-R-001, EMU checkout in Orbiter Y1103-02, EMU Performance Checks.</p> <p>F. Operational Use - Crew Response - PreEVA: Troubleshoot problem, if no success, consider third EMU if available. Otherwise, EMU go for EVA prep on battery power. Consider use of spare battery for in-bulk battery swap prior to EVA. PostEVA: Remain on battery power until EMU doffed. Training - Standard training covers this failure mode. Operational Considerations - EVA checklist procedures verify hardware integrity and systems operational status prior to EVA. Flight rules define go/no go criteria related to SCU power.</p>