

FAILURE MODE EFFECTS ANALYSIS/CRITICAL ITEMS LIST**FMEA NUMBER: CSD-TB-34****ORIGINATOR: JSC****PROJECT: EDFT-02****PART NAME: EVA C/O MECH****LRU/ORU PART NUMBER: SED39127468-30/ QUANTITY: 1****PART NUMBER: SED39127549-301****LRU/ORU PART NAME: ORT****SYSTEM: DTO 671****LSC CONTROL NO: N/A****DRAWING/REF DESIGNATOR: SEE P/N****SUBSYSTEM: EVA****ZONE/LOCATION: CABIN/PLB****EFFECTIVITY/AFFECT STAGE: STS-69 & SUBS****CRITICALITY:****CRITICAL ITEM: Yes****SUCCESS PATHS: 2****CRITICALITY CATEGORY: 1R/2****SUCCESS PATH REMAINING: 1****REDUNDANCY SCREENS:****A/1. C/O PRELAUNCH: Pass****2. C/O ON ORBIT: N/A for NSTS****B/3. DETECTION FLIGHT CREW: Pass****4. DETECTION GROUND CREW: N/A****C/5. LOSS OF REDUNDANCY FROM SINGLE CAUSE: Pass**

FUNCTION: The ORU Restraint Tether (ORT) is a semi-rigid restraint aid available to EVA crewman to temporarily mount ORU's. The ORT interfaces with an ORU's Tether loop utilizing a push lock tether tool or nested hook tether tool and will be mounted to structure through an EVA change out mechanism. The base plate the ORT mounts to will incorporate EVA contingency bolts on STS-69. The closed coupled configuration ORT is a modified version of the ORT. It is the same as the ORT without the ball stack. The closed coupled configuration ORT interfaces at it's base to a load alleviating gimbal. All ORT configurations are designed to restrain an ORU up to 50 lb. mass.

FAILURE MODE CODE: N/A for NSTS**FAILURE MODE: EVA changeout mechanism jams.****CAUSE: Contamination, galling, piece part defect****REMAINING PATHS: EVA contingency bolts or jettison of TERA or PFRWS****EFFECT/ MISSION PHASE: EVA****CORRECTIVE ACTION: Remove bolts holding ORT base to structure (Task Board on STS-69) or jettison attached equipment.****-FAILURE EFFECTS-****END ITEM/LRU/ORU/ASSEMBLY:**

Unable to release ORT from ORT base.

SUBSYSTEM/NEXT ASSEMBLY/INTERFACE: N/A**SYSTEM/END ITEM/MISSION: Partial loss of DTO objectives.****CREW/VEHICLE : None for single failure, possible damage to vehicle if ORT is not properly stowed for landing.**

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PART NAME: EVA C/D MECH	LRU/ORU PART NUMBER: SED39127468	QUANTITY: 1
PART NUMBER: SED39127549-301	LRU/ORU PART NAME: ORT	SYSTEM: DTO 671
LSC CONTROL NO: N/A	DRAWING/REF DESIGNATOR: SEE P/N	SUBSYSTEM: EVA
ZONE/LOCATION: CABIN/PLB	EFFECTIVITY/AFFECT STAGE: STS-69 & SUBS	

HAZARD INFORMATION:

HAZARD: YES: x NO:

HAZARD ORGANIZATION CODE: N/A

HAZARD NUMBER: N/A

TIME TO EFFECT: Hours
 TIME TO DETECT: Seconds
 TIME TO CORRECT: Minutes
 FAILURE DETECTION/FLIGHT: Visual

REMARKS:

-RATIONALE FOR ACCEPTABILITY-

(A) DESIGN: The ORT Assy. is designed to withstand a 1800 in-lb bending and torsion, 75 lb. tension load along its long axis, and 125 lb. in shear without functional degradation. The sheath is designed to withstand a 75 lb. load independently from the cable in the ball stack. If cable failure were to occur, the sheath is designed to prevent any of the loss parts from becoming free in the PLB. Load verification is performed by analysis with a ultimate factor of safety of 2.0. Positive margin is required. For structural items the ORT is attached to that can not be jettisoned, a EVA releasable interface to that structure is required.

(B) TEST: Applicable requirements from JSC-33034.

Acceptance:

- 1) Functional: Verified at Predelivery Acceptance Test, Preinstallation acceptance, and Pre/Post environmental test. Minimum of 30 actuation cycles total.
 - a) EVA release mechanism installation and removal force shall be between 2 and 10 lb. verified per TPS.
 - b) The torque required to operate the mechanism shall be between 1 and 5 in-lb. and the force required to actuate the mechanism shall be between 1 and 5 lb. verified per TPS.
 - c) 75 lb. tensile load test performed on the cable at PDA verified per TPS.
 - d) 125 lb. tensile load test performed on the sheath at PDA verified per TPS.
- 2) Environmental: Acceptance Vibration
 The ORT is subjected to the following vibration in each axis for a duration of 1 minute verified per TPS:

20 Hz	0.01 g ² /Hz
20 to 80 Hz	+3.0dB/oct
80 to 350 Hz	.040 G ² /Hz
350 to 2000 Hz	-3.0dB/oct
2000 Hz	0.007 g ² /Hz

load factor 6.1 Grms

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LSC CONTROL NO: N/A	DRAWING/REF DESIGNATOR: SEE P/N	SUBSYSTEM: EVA
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Qualification:**Environmental:**

Thermal: Functional verification performed at -100°F and +200°F. Forces are subjectively evaluated during the test and measured pre and post test verified per TPS.

Load test: Sheath load tested to 183 lb, verified per TPS.

(C) INSPECTION:

Fabrication - All ORT components are verified to be built to print and generally clean individually. The ORT assy. is verified to be visually clean at predelivery acceptance.

Test - Quality Assurance surveillance is required at all test and inspections. Discrepancy reports are written on all noncompliance's.

(D) FAILURE HISTORY: None**(E) OPERATIONAL USE:**

- 1) **Operational Effect -** Unable to remove ORT from task board or PLTT from ORT.
- 2) **Crew Action -** Remove contingency release bolts or jettison equipment.
- 3) **Crew Training -** Crew trained in proper operation of ORT.
- 4) **Mission constraint -** When rigidified the ORT should not be operated with the bail stack bent at more than 120 degrees.
- 5) **In Flight Checkout -** The EVA change out mechanism operation will be functional verified at the of operation during the EVA.

(F) MAINTAINABILITY: N/A**PREPARED BY: G.Wright****REVISION:****DATE: 5/15/95****WAIVER NUMBER**