REVISED 5-7-87 UNIT Cable FMEA NO. H 4.26.2 SHUTTLE CCTV 2793287-503 DWG NOT CRITICAL ITEMS LIST ESSUED 10- 4-86 CRITICALITY 2/1R SHEET FATLURE MODE AND FAILURE EFFECT CAUSE ON END LYEN RATIONALE FOR ACCEPTANCE Loss of Limit Up SW Tilt will not move in DESIGN FEATURES up direction (CCW). Short to GND The W4 PTU cable is a 44-inch long, 25-wire assembly terminated by 37 pin connectors at each end. The video and sync/cmd wires are shielded Indian shielded and twisted pairs Horst Case: of #24 wire. The cable connects the TVC and PTU. Connector types KJ66814M35SM16 have Filt will not move to up been selected. direction (CCW) for elbow The cable design is taken from the successfully flown Apollo program. The design is a Camera to permit arm stowcable-connector assembly in which the wire terminations are protected from excessive age. flexture at the joint between the wire and the connector terminal. The load concentration is moved away from the conductor connection and distributed axially along the length of the conductors encapsulated in a potted-taper profile. This technique also protects the assembly from dirt and entrapped moisture which could cause problems In space. The cable and its components meet the applicable requirements of MASA, Military and RCA specifications. These requirements include: General/Hechanical/Electrical Features Design and Construction Materials Terminal Solderability Environmental Qualification. Marking and Serialization Traceability and Occumentation

REVISED 5-7-87 TINTE CabTé 2293287-503 FMEA NO. H 4.26.2 SHUTTLE CCTV DWG NO. [SSUED | 10-14-86 CRITICAL TIEMS LIST SHEET 2/18 CRITICALITY FATLURE HOUE AND FATEURE EFFECT ON END ITEM CAUSE RATIONALE FOR ACCEPTANCE Loss of Limit Up SW OUAL SEICATION TEST Tilt will not move in up direction (CCN). Qualified by 1.) similarity to previous successful space programs and 2.) by use during Short to GND qualification tests of CCTV LRUs. Worst Case: Tilt will not move in up ACCEPTANCE TEST direction (CCN) for elbow The cable acceptance test consists of an olymmeter check to assure that each wire camera to permit arm stowconnection is present and intact. Results are recorded on data sheets. age. OPERATIONAL TEST The following tests verify that CCTV components are operable and that the commands from the PNS (A7AT) panel switch, through the RCB, through the sync lines to the Camera/PTU. to the Camera/PTH command decoder are proper. The tests also verify the camera's ability to produce video, the VSU's ability to route video and the monitor's ability to display video. A similar test verifies the MDM command path. Pre-Launch on Orbiter Test/In-Flight Test Power CCTY System. Select a monitor yia the PHS panel, as destination and the camera under test as source. Send "Camera Power On" command from PMS panel. Select "External Sync" on monitor. Observe video displayed on monitor. [f video on munitur is synchronized [i.e., stable raster), then this indicates that the camera is receiving composite sync from the RCM and that the camera is producing synchronized video. Send Pag. Tilt, Focus, Zoom, ALC, and Gamma commands and visually (either via the monitor or direct observation) verify proper operation. Select Downlink as destination and camera under test as source. Observe video routed to downlink. Send "Camera Power Off" command via PHS panel. Repeat Steps 3 through 9 except issue commands via the MOM command path. This proves that the CCTV equipment is operational if video is satisfactory.

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FMEA NO. <u>W 4.26.2</u> CRITICALUTY <u>2/IR</u>	-	SHUTTLE CCTV CRITICAL ETENS LIST	I SHEET 3 0F 5	
FATLURE MODE AND	FAILURE EFFECT ON END ITEM	NATIONALE FOR ACCEPTANCE		
CAUSE Loss of Limit Up SW Short to GND	Yilt will not move in up direction (CCW). Morst Case: Tilt will not move in up direction (CCW) for elbow camera to permit arm stowage.	Procurement Control - Wire, connectors, solder, etc. ar and suppliers which meet the requirements set forth in Plan Work Statement (WS-2593176). Incoming Inspection & Sturage - Incoming Quality inspectations and parts. Headlits are recorded by lot and a control numbers for future reference and traceability. Naterial Controlled Stores and retained under specified fabrication is required. Non-conforming materials are (MR8) disposition. (PAI-307, PAI IQC-53). Assembly & Test - Prior to the start of assembly, all by stock room personnel as the items are accumulated to verified again by the operator who assembles the kit by as-built-parts-list (ABPL). Specific instructions are given in assembly drawing not called out in the Fabrication Procedure and Record (FPI Process Standard criuping flight connector contacts, 2; splicing of standard interconnecting wire using Raycher Process Standard marking of parts or assemblies with elemetrial and test procedure (TP-AT-2293287). Quality at the completion of key operations. Preparation for Shipment - When fabrication and test in packaged according to 2280746, Process Standard for PaA11 related documentation including assembly drawings.	AATIONALE FOR ACCEPTANCE IION Int Control - Wire, connectors, solder, etc. are procured from approved vendors fers which meet the requirements set forth in the CCTV contract and Quality Statement [WS-2593176]. Inspection & Storage - Incoming Quality inspections are made on all received and parts. Hegults are recorded by lot and retained in file by drawing and unders for future reference and traceability. Accepted Items are delivered to Controlled Stores and retained under specified conditions until cable on is required. Non-conforming materials are held for Material Review Board position. (PAI-307, PAI IQC-53]. § Test - Prior to the start of assembly, all items are verified to be correct room personnel as the items are accumulated to form a kit. The items are again by the operator who assembles the kit by checking against the parts-list (ABPL). instructions are given in assembly drawing notes and applicable documents to in the Fabrication Procedure and Record (FPR-2293287). These are 2280800 - tandard crimping flight connector contacts, 2280801 - Process Standard in-line of standard interconnecting wire using Raychem solder sleeves, 2280876 - tandard interconnecting wire using Raychem solder sleeves, 2280876. Potting and test procedure (TP-AT-2293287). Quality and BCAS Inspections are performed appletion of key operations. on for Shipment - When fabrication and test is complete, the cable assembly is according to 2280746, Process Standard for Pactaging and Handling Guidelines, ed documentation including assembly drawings, Parts List, ABPL, Test Data, etc. and held in a documentation folder assigned specifically to each cable	

RFV]SEO 5-7-87 UNIT Cable W 4.26.2 SHUTTLE CCTV CRITICAL (TEMS LIST DHE NO. 2293287-503 FMEA MO. 1\$5UED 10-14-B5 SHEET CRITICALITY 2/18 FAILURE EFFECT ON END ITEM FAILURE MODE AND RATIONALE FOR ACCEPTANCE CAUSE Tilt will not move in Loss of Limit Up SW FAILURE HISTORY up direction (CCW). There have been no reported failures during RCA testing, pre-flight or flight. Short to GMD Worst Case: Tilt will not move in up direction (CCN) for elbow camera to permit arm stowage.

REVISEU 5-7-87 UNIT Cable 2293287-503 DNE NO. SHOTTLE CCTV N 4.26.2 FMEA ND. ISSUED TO-14-86 CRITICAL TIEMS LIST SHEET 2/18 CRITICALITY FATEURE EFFECT TATELINE HORE AND RATIONALE FOR ACCEPTANCE ON END ITEM CALISE OPERATIONAL EFFECTS Tilt will not move in Loss of Limit Up SW up direction (CCN). Loss of ability to position the Elbow camers. Possible inability to stow the RMS if the Short to GMD Worst Case: elbow camera physically interferes with a payload. If RMS cannot be stowed the port payload may door cannot be closed. Loss of crew and vehicle. Tilt will not move in up direction (CCV) for elbow camera to permit arm stuw-CREW ACTIONS aye. Perform EVA to reposition the elbow camera, use RNS motion to reposition the camera, or jettison the RMS. CREW TRAINING Crew should be trained in contingency EVA and RMS operations procedures. MISSION CONSTRAINT Do not manifest fibuw camera for any flight where the payload and the elbow camera can interfere with each other (for any pan or tilt angle). If the comera must be flown do not change the camera position until the interfering payload is deployed.